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# LANDSAT 7 SYSTEM

# DATA FORMAT CONTROL BOOK (DFCB) VOLUME IV – WIDEBAND DATA

26 February 1998 Revision H

Contract No. NAS5-32633

Prepared for:

NASA/Goddard Space Flight Center Landsat Project Office Code 430 Greenbelt, MD 20771

Prepared by:

Lockheed Martin Missiles & Space P.O. Box 8555 Philadelphia, PA 19101

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**FOR** 

**LANDSAT 7 SYSTEM** 

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Prepared by: [CAGE Code: 79272]

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#### SECTION 1 SCOPE

#### 1.1 IDENTIFICATION

This volume defines the formats used for the transmission of Landsat 7 wideband data to the Ground Segment and to the International Ground Stations.

#### 1.2 SYSTEM OVERVIEW

The Wideband Mission Data consists of all Enhanced Thematic Mapper Plus (ETM+) instrument data with embedded Payload Correction Data (PCD). The PCD is the ancillary spacecraft data needed to properly process the ETM+ image data, and is provided to the ETM+ by the satellite Command and Data Handling Subsystem. The Wideband ETM+ data is transmitted to the ground station at Sioux Falls, South Dakota or to the International Ground Stations (IGS) in real-time or subsequent playback from solid state recorders using 3 X-Band antennas. One of the three X-Band antennas can transmit data over one, two, or three frequencies at a rate of 150 Mbps per frequency. The second transmits over one or two frequencies. The third X-Band antenna operates at a single frequency with a rate of 150 Mbps.

## **SECTION 2** APPLICABLE DOCUMENTS

#### 2.1 **GOVERNMENT DOCUMENTS**

The following documents of the exact issue shown, form a part of this specification to the extent specified herein. In the event of a conflict between the documents referenced herein and this document, the contents of this document shall be considered a superseding requirement.

#### Specifications:

Military Standard, Sixteen-Bit Computer Instruction Set Architecture MIL-STD-1750

#### 2.2 NON-GOVERNMENT DOCUMENTS

The following documents, of the exact issue shown where listed or the latest approved issue where not listed, form a part of this document to the extent specified herein. In the event of a conflict between the documents referenced herein and this document, the contents of this document shall be considered a superseding requirement.

#### Specifications:

PS230020620 Critical Item Product Function Specification for the Landsat 7 Payload

Data Formatter 7 July 1995

Standards:

CCSDS 701.0-B-1 Advanced Orbiting Systems, Networks and

Data Links: Architectural Specification Issue 1

October 1989

CCSDS 101.0-B-3 **Telemetry Channel Coding** 

May 1992

Other Publications:

PS23007610 Program Coordinates System Standard

#### SECTION 3 IMPLEMENTATION

#### 3.1 WIDEBAND MISSION DATA PROTOCOL

The Landsat 7 System requirements specify that all space to ground data transfers utilize the Consultative Committee for Space Data Systems (CCSDS) telecommand and telemetry recommendations. The CCSDS protocol that will be used on Landsat 7 can be divided into three layers. The three layers in the data structure roughly correspond to the Open System Interconnection (OSI) reference model as shown below in Figure 1. They are:

- 1. The Application Data Layer which consists of the actual mission data and associated Error Detection and Correction (EDAC) codes.
- 2. The Virtual Channel Layer which consists of CCSDS headers and trailers required for end-to-end flow and error control, and synchronization markers.
- 3. The Physical Channel Layer which is responsible for transmission of the data across a physical medium; in the case of Landsat 7 the medium is a QPSK modulated X-band radio channel.

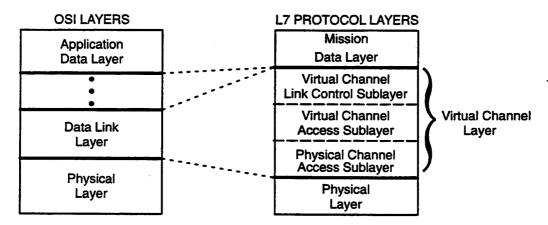


Figure 1. L7 Protocol Layering Relationships

The wideband communications protocol structure is compliant with a CCSDS Grade 3 Bitstream service (see Figure 2.) The application or mission data is delivered to the protocol function in a bitwise serial (bitstream) manner at the Virtual Channel Link Control (VCLC) sublayer. A Landsat 7 unique data pointer is inserted to provide the end user the ability to accurately reconstruct the data stream. Additionally, a BCH EDAC field is added to achieve the required BER of 10<sup>-6</sup>. From a protocol perspective, the BCH EDAC field can be considered part of the data field.

The data stream is subsequently fed into the Virtual Channel Access (VCA) sublayer where the appropriate headers and trailers are appended to provide for end-to-end flow control. The result of this process is the creation of a Virtual Channel Data Unit (VCDU). Just prior to transmission on the Physical Channel, a synchronization header is added to create a Channel Access Data Unit (CADU). Finally, the data stream is modulated and transmitted over the physical channel.

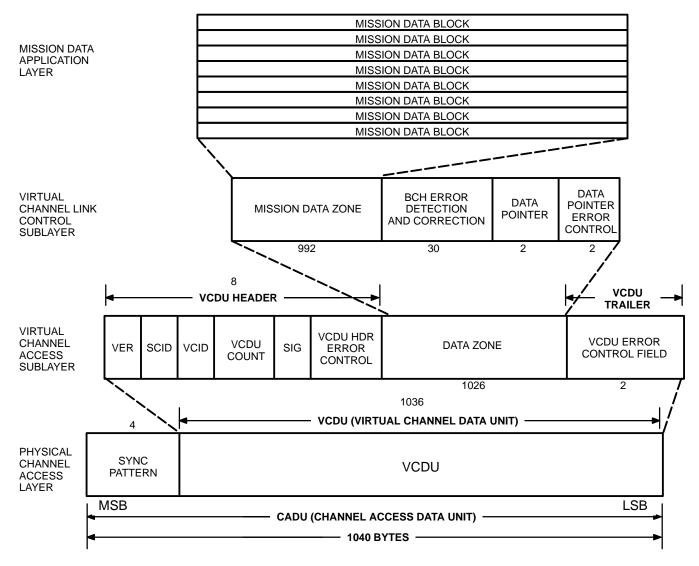


Figure 2. L7 Wideband Protocol Layers

For each of the data fields addressed in this Section 3, the following conventions are used:

- 1. The most significant bit is transmitted first.
- 2. The most significant bit in a data field is labeled bit 0.
- 3. The size of a data word is 8 bits.

#### 3.1.1 MISSION DATA LAYER

The application layer consists of the mission data streams for the ETM+ payload. The data streams consist of 992 bit blocks of payload data. Detailed information on the formats of the mission data is provided in sections 3.2 and 3.3 of this volume.

#### 3.1.2 VIRTUAL CHANNEL LAYER

The virtual channel layer functional responsibilities can be divided into three categories: bitstream segmentation with error detection and correction; providing headers and trailers as needed for

end-to-end flow control; providing synchronization for transfer across the Physical Channel. These functions are provided by the Virtual Channel Link Control sublayer (VCLC) and the Virtual Channel Access (VCA) sublayer, and the Physical Channel Access (PCA) sublayer, respectively. The most significant bit is transmitted first.

## 3.1.2.1 Virtual Channel Link Control Sublayer

The VCLC is responsible for segmenting the bitstream ETM+ data into a predefined zone of 7936 bits. The VCLC is responsible for providing error protection of the data zone. This will be accomplished by a 240 bit field containing a BCH EDAC field, and appended to the end of the mission data. A data pointer, defined below, is attached immediately following the mission data BCH code. The pointer is protected by a 16 bit field containing a BCH EDAC field as defined in section 3.1.2.1.1.1. The data pointer error control EDAC field is inserted following the data pointer, as shown in Figure 3.

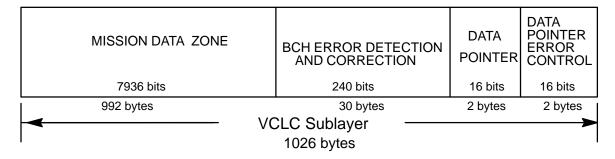


Figure 3. VCLC Sublayer

#### 3.1.2.1.1 Data Pointer

The data pointer structure is as shown below in Figure 4. The data pointer field contains a synchronization pointer which correlates the transfer frame timing to the Minor Frame timing of the ETM+ data. The length of the pointer field is 16 bits. The 7 least significant bits are used to indicate the number of data words (0 to 84) between the start of the data zone and the first word of the first full minor frame of the VCDU data zone. The 9 most significant bits are each set to "0". The most significant bit is transmitted first. The data pointer is always valid except for the situation described in Section 3.2.6.

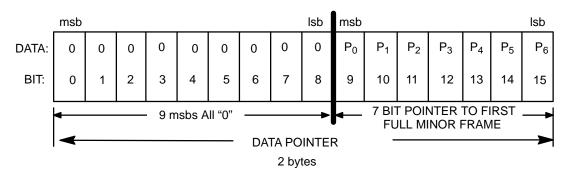


Figure 4. Data Pointer Bit Field Definition

#### 3.1.2.1.2 Data Pointer Error Control

The Data Pointer field must have a separate error protection scheme than the mission data zone. The error protection will be accomplished using a (31,16,3) BCH code, which provides the capability to correct three (3) errors in 16 data bits and 15 code bits. This code is generated using the polynomial:

$$g(x) = x^{15} + x^{11} + x^{10} + x^9 + x^8 + x^7 + x^5 + x^3 + x^2 + x + 1$$

The code bits will be located in the 15 least significant bits of a two byte field immediately following the Data Pointer. In order to preserve the CCSDS eight bit field length requirement, a fill bit will be inserted at the beginning of the 15 code bits as shown in Figure 5. The value of the fill bit will be set to "0" at all times. Since the fill bit is "0", the BCH generator will not be affected.

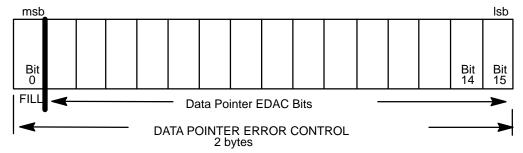


Figure 5. Data Pointer Error Control Bit Field Definition

#### 3.1.2.1.3 Mission Data Zone

The Mission Data Zone is a 7936 bit (992 bytes) field that contains image or DC restore/calibration data of the ETM+.

#### 3.1.2.1.3.1 Mission Data BCH Error Control

The mission data will be protected with a (1023, 993, 3) BCH code. The generator polynomial for this code:

$$g(x) = x^{30} + x^{28} + x^{23} + x^{21} + x^{19} + x^{16} + x^{12} + x^8 + x^4 + x + 1$$

This code has the capability to correct three errors in 993 data bits and 30 code bits. Since the mission data field consists of 8 blocks of 992 bits, a fill bit must precede each 992 bit block of mission data to correctly employ the BCH algorithm. The fill bit **will not** be transmitted. The value of the fill bit will always be "0". Proper error detection and correction by the ground processing software will require the insertion of a filler bit at the beginning of each 992 bit block prior to execution of the BCH code or will require preloading the intermediate result of the BCH code that is calculated from applying the preceding fill bit to the BCH algorithm.

BCH algorithm will generate 30 code bits for each of the eight 992 bit blocks of mission data. The 8 sets of 30 code bits will be appended to the end of the mission data field in a predefined 240 bit field. The first 8 bits will be the code bits corresponding to the first code bit in each of the 8 BCH encoders. Bits 8–15 will correspond to the second code bit in each of the 8 BCH encoders in the mission data zone, and so forth with the final 8 bits corresponding to the 30th, and final, check bit for each of the 8 BCH blocks. Figure 6 illustrates this relationship. Figure 7 illustrates the BCH encoding circuit. It should be noted that the data transferred out of the encoder is not interleaved, however, the code bits are interleaved.

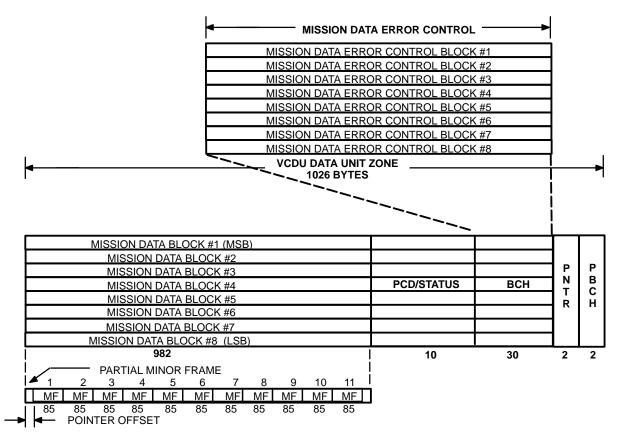


Figure 6. Mission Data Error Control BCH Code Block Relationships

#### 3.1.2.2 Virtual Channel Access Sublayer

The VCA sublayer is responsible for appending the appropriate headers and trailers to the data stream to provide for end-to-end flow control. This is accomplished by inserting the 64 bit (eight byte) VCDU primary header at the start of the data stream, and appending the 16 bit (2 byte) VCDU Error Control Field at the end of the data stream.

#### 3.1.2.2.1 VCDU Primary Header

The VCDU Primary header will contain 64 bits (eight bytes) due to the fact that the optional Header Error Control field has been included. The remaining fields are defined below and illustrated in Figure 8.

#### **3.1.2.2.1.1 Version Number**

The first two bits of the Primary header are reserved for the Version number of the CCSDS packet. The two Version bits will be set to 01<sub>2</sub>, indicating a Version–2 CCSDS Packet.

#### 3.1.2.2.1.2 Spacecraft Identifier

Bits 2 (two) through 9 (nine) are reserved for the Spacecraft Identifier. The Spacecraft ID is chosen by the Secretariat of the CCSDS. The value that has been assigned for Landsat 7 is 00010101<sub>2</sub>.

#### 3.1.2.2.1.3 Virtual Channel Identifier

Bits 10 through 15 of the Primary header are reserved for the Virtual Channel Identifier (VCID). The VCID field allows up to 64 virtual channels to run concurrently on one physical channel. For Landsat 7,

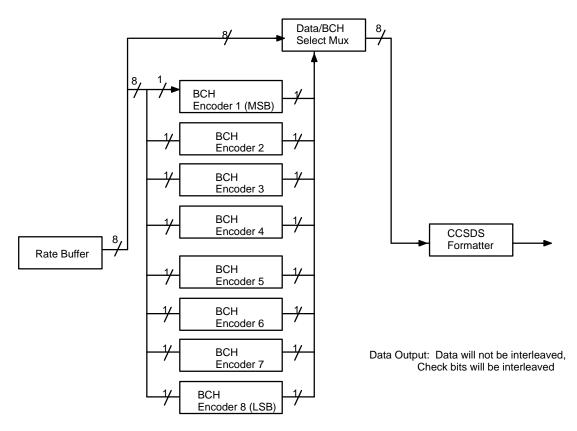


Figure 7. BCH Encoding

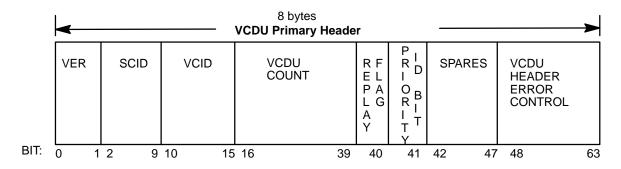


Figure 8. VCDU Primary Header

the VCID field will be used to identify from which channel the subject data was generated. For Landsat 7, Channel 1 identifies Format 1 data: Bands 1 through 5, and IR Band 6. Channel 2 identifies Format 2 data: IR Bands 6 and 7, and the Pan Band. The values will be as follows:

- 1. ETM+ Data Format 1 000001<sub>2</sub>
- 2. ETM+ Data Format  $2 000010_2$

#### 3.1.2.2.1.4 Virtual Channel Data Unit Counter

Bits 16 through 39 are reserved for the VCDU Counter field. The 24 bit field will be used in conjunction with the VCID to maintain a separate counter for each instrument channel. The first count will be zero and the counter will be reset to zero each time an instrument transitions from "standby" to

"on" (ETM+). The counter will increment once for each VCDU. The counter will roll over when it exceeds 16,777,215, irregardless of whether this occurs on scene boundaries or in the middle of a current scene.

#### **3.1.2.2.1.5** Replay Flag

Bit 40 of the VCDU Primary header is reserved for the Replay Flag Bit. The Replay Flag will not be used in Landsat 7; therefore the value of this bit will be set to "0" for all VCDU's.

## 3.1.2.2.1.6 Priority Identification Bit

Bit 41 of the VCDU header will be the Priority Identification Bit. This implementation is unique to Landsat 7. The Priority ID bit will be used to differentiate mission data that the system identifies as requiring immediate attention by ground processing. The implementation of this bit will be as follows:

- 1. Priority Data − 1<sub>2</sub>
- 2. Routine Data  $-0_2$

#### 3.1.2.2.1.7 Reserved Spares Field

Bits 42 through 47 are reserved for future applications by CCSDS. They will be set to the value of all zeros as follows:

1. Reserved Spares – 000000<sub>2</sub>

#### 3.1.2.2.1.8 VCDU Header Error Control Field

The optional EDAC to protect the header will be utilized for Landsat 7. Bits 48 through 63 will be the check symbols of a shortened Reed-Solomon (10,6) code and further described in Appendix B. The code protects all fields within the header with the exception of the VCDU counter. Consequently, the value of the VCDU header check symbols are known for a given instrument channel and data priority; (i.e., they do not need to be computed dynamically on the S/C but will be computed on the ground and checked for errors). The values for these fields will be defined as follows:

- 1. ETM+ Data Format 1 Priority 6594<sub>16</sub>
- 2. ETM+ Data Format 1 Routine BF82<sub>16</sub>
- 3. ETM+ Data Format 2 Priority 03A5<sub>16</sub>
- 4. ETM+ Data Format 2 Routine D9B3<sub>16</sub>

#### 3.1.2.2.2 *VCDU Trailer*

The VCDU Trailer is generally an optional component of the VCDU. However, CCSDS mandates that the VCDU Error Control Field be present within any channels that are not Reed-Solomon encoded. The Operational Control Field option will not be used.

#### 3.1.2.2.2.1 VCDU Trailer Error Control Field

The VCDU Trailer Error Control Field is a 16-bit CRC code which provides the capability to detect errors that may have been introduced during data transmission. The last 16 bits of the VCDU,

specifically, bits 8272 through 8287, will contain the 16 generated parity symbols. The generator polynomial is:

$$g(x)=x^{16}+x^{12}+x^5+1$$
.

Both the encoder and decoder are initialized to the "all ones" state for each VCDU. The parity generation is performed over the entire data space of the VCDU, excluding the 16 bits of error control. The generated parity symbols are inserted into the VCDU error control field, which occupies the last 16 bits of the VCDU.

#### 3.1.2.3 Physical Channel Access Sublayer

The Physical Channel Access Sublayer is responsible for generating a continuous and contiguous stream of serial data. The resulting data unit is known as a Channel Access Data Unit (CADU). The Landsat 7 Wideband CADU is shown in Figure 9.

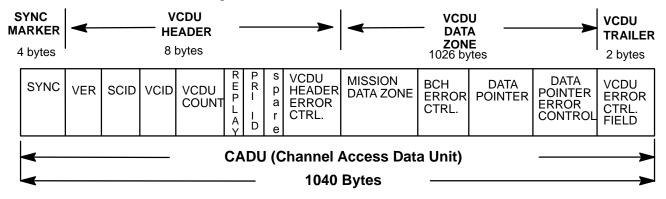


Figure 9. Channel Access Data Unit Description

#### 3.1.2.3.1 Pseudo-Randomizer

The use of the pseudo-randomizer is necessary to guarantee the bit transition density required to maintain bit synchronization with the received signal. The method for ensuring sufficient transitions is to exclusive-OR each bit of the VCDU (does not include the Sync field) with a standard pseudo-random sequence. On the receiving end, the same sequence is exclusive-ORed with the received VCDU to remove the randomized pattern and restore the original data. The generator polynomial, logic diagram, and associated usage constraints can be found in CCSDS 101.0–B–3, paragraphs 6.3, 6.4 and 6.5.

#### 3.1.2.3.2 Synchronization Marker

A CADU is delineated by a synchronization marker. CCSDS defines this synchronization marker to be a 32 bit field located at the beginning of each CADU. The most significant bit is transmitted first. The synchronization marker will have the following value:

1. CADU Synchronization Marker – (msb) **1ACFFC1D**<sub>16</sub> (lsb)

#### 3.1.3 PHYSICAL CHANNEL LAYER

The Physical Channel layer provides the medium for transmission of the CADU's between the space vehicle and the ground. Details regarding the data modulation formats can be found in the RF ICD Between Landsat 7 and the Space Network (SN), Ground Network (GN), and Landsat 7 Ground Station (LGS), 23007638.

#### 3.2 ETM+ MISSION DATA FORMATS

The Enhanced Thematic Mapper Plus (ETM+) is a payload on the Landsat 7 space vehicle. The ETM+ payload contains sensors that detect earth scene radiation in visible and near infrared (VNIR) bands, short wavelength infrared (SWIR) bands, and thermal long wavelength infrared (LWIR) band. ETM+ image data are stuffed into most of the 7936 bit slots in CADU's as defined in Section 3.1 of this volume. The structure of the ETM+ data formats within a given scene for all data modes are the subject of this section.

#### 3.2.1 ETM+ DATA FLOW AND FORMATTING

Each of the two multiplexers receives all 136 analog inputs (one per detector) from focal planes. These inputs are separated into two groups; 88 inputs containing Bands 1–6 and 56 inputs containing Bands 6, 7, and Pan. The format of the focal planes, Instantaneous Field of View (IFOV), position and orientation of the detectors are described in Section 3.2.2. Each of the multiplexers has two high speed outputs that simultaneously output both scene data formats such that one activated multiplexer provides for the required ETM+ output data while the other multiplexer remains in an unpowered standby mode as selected by external command. Multiplexer outputs will be combined as shown in Figure 10 to provide two high speed data outputs. The A/D converter output within each multiplexer provides digital data that is transferred to the Minor Frame Formatters. The Minor Frame formatters format all of the data into the two scene data formats described in Section 3.2.5.3 which have the minor frame structure described in Section 3.2.4. After construction of the minor frames, the PCD/Status words are added (Section 3.2.6), the data is BCH encoded, and CCSDS formatted. The PCD data types and formats are described in Section 3.2.7. The description of the BCH encoding and CCSDS formatting are provided in Section 3.1. Figure 10 provides a functional diagram of the data flow and formatting.

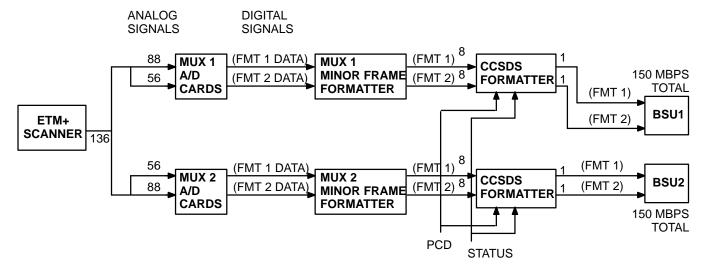


Figure 10. ETM+ Data Flow and Formatting

#### 3.2.2 FOCAL PLANE, SPECTRAL BANDS, IFOV SIZE, AND GROUND RESOLUTION

ETM+ collects, filters, and detects radiation from the earth in a swath 185 km wide as it passes overhead and provides the necessary cross track scanning motion while the spacecraft orbital motion provides an along-track scan. The ETM+scanner contains two focal plane assemblies. The primary focal plane assembly including optical filters, detectors, and preamplifiers for five of the eight ETM+ spectral bands (Bands 1–4, 8). The second focal plane assembly is the cold focal plane assembly. This includes optical

filters, infrared detectors, and input stages for the remaining three ETM+ spectral bands (Bands 5, 6, and 7). When the ETM+ focal plane and scan pattern are projected on the ground (Figure 11), detector No. 1 of each band appears at the leading, Southern edge of the scan line. At each instance of time, the odd detectors are to the East and the even detectors to the west and shifted North relative to the odd detectors by the distance of one (1) detector (pixel) length in that band.

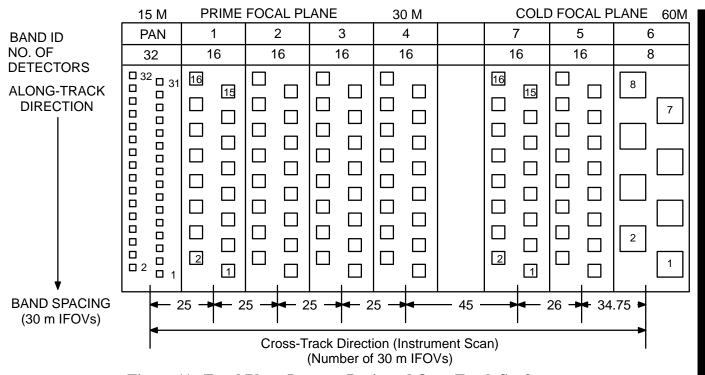


Figure 11. Focal Plane Layouts Projected Onto Earth Surface

#### 3.2.3 ETM+ MAJOR FRAME STRUCTURE

The ETM+ data stream is composed of a continuous succession of Major Frames. For nominal operations on orbit and in vacuum each Major Frame contains the data for an entire period of one complete scan of the ETM+ scan mirror (either direction). A Major Frame period  $\sim 71.820$  ms includes not only the period during a scan but also the turnaround interval (including calibration data) when the scan mirror changes direction for the next scan. ETM+ Major frames are partitioned into Minor Frames (~7473 Minor Frames per Major Frame). During both Instrument and Observatory level I&T, most testing will be conducted at ambient temperature and pressure, which increases the turnaround interval so that the Major Frame period becomes 71.993 ms with  $\sim$  7491 Minor Frames per Major Frame. It should be noted: (a) these numbers are nominal, and, (b) turn-around-time will begin to increase after a couple of years if wear occurs on-orbit as happened with Landsat 5. Turn-around-time can be trended with the wideband data by using the time code data (section 3.2.5.2) and the active scan time (section 3.2.5.5). The first Minor Frame of the Major Frame is initiated by a mechanically/optically timed pulse. This Minor Frame is called the Line Sync Code or Scan Line Start (SLS). The six Minor Frames immediately following the SLS Minor Frame describe the spacecraft time (Time Code). Scene Data (partitioned into Minor Frames) occupies the Major Frame until another mechanically/optically timed mirror pulse preempts the scene data with an End-of-Line (EOL) Code. This code signals the end of an active image scan. Shortly following the EOL code, the Scan Line Length (SLL) and Scan Direction (SD) codes occur (Summarized as Scan Line Data in Section 3.2.5.5). The data provided in both the

Scan Line Length and the Scan Direction codes are descriptions of the scan prior to the current scan. Figure 12 describes the relationship between major frames, minor frames, and VCDUs.

#### 3.2.4 MINOR FRAME DATA STRUCTURE

The Minor Frame Data Structure is a specific pattern for organizing groups of ETM+ data words. This pattern is based on the architecture of the L–7 Auxiliary Electronics Module (AEM) hardware that samples, digitizes, and groups analog video signals from the ETM+ scanner to form scene data. The Minor Frame Data structure is 85 words (8 bits) in length consisting of 16 separate groups of 5 words, 4 data words from Band 6, and 1 spare data word. Band 6 data shall alternate between the odd and even pixels for each successive minor frame and shall be resynchronized to odd pixel data for the first minor frame (line sync code) of each new scan. Each bit of the spare data word is set to zero. The Minor Frame Data structure is shown in Figure 13. Note that the odd numbered words are transmitted first followed by the even numbered words and then Band 6 data. Format 1 contains Band 6 low gain and Format 2 Band 6 high gain.

#### 3.2.5 MINOR FRAME FORMATS

Seven separate minor frame formats or minor frame format groups are used to accommodate the required ETM+ scan events and scene data. The minor frame formats and data content are described in the following paragraphs. Note that the tables describing the formats are shown in numeric word sequence and not in time sequence.

#### 3.2.5.1 Line Sync Code Format (Scan Line Start)

The L–7 AEM asynchronously detects the occurrence of the trailing edge line start pulse from the scan mirror assembly. Upon detection of the start pulse, a new scan line is initiated which is synchronized to the start of the next data word. The Line Sync Code is generated by the L–7 AEM at the beginning of each new scan line and inserted into minor frame zero and pre-empts all minor frame video except Band 6 data. However, Band 6 data is not valid for minor frame zero. In addition, each of the 16 groups of 5 words within a given minor frame will contain a single bit binary value (0 or 1) that is replicated for all of the bits in the group (40 bits per group). Table 1 lists the single bit values for the Line Sync Code.

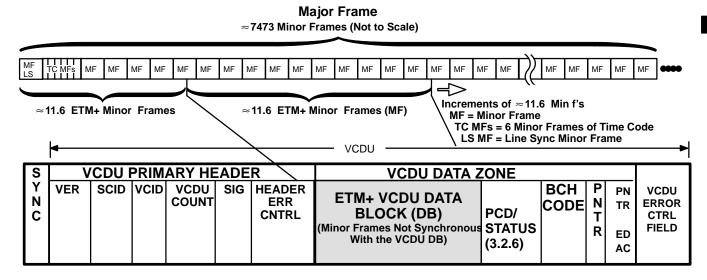


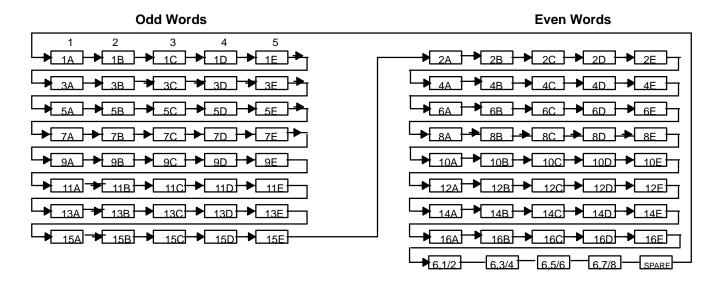
Figure 12. ETM+ Major Frame, Minor Frames, CADU

TABLE 1. LINE SYNC CODE (LSC) FORMAT (1 MINOR FRAME)

1A-E	2A-E	3А-Е	4A–E	5A-E	6A-E	7A-E	8A-E	9A-E	10A-E	11A-E	12A-E	13A-E	14A-E	15A-E	16A-E
"1"	"0"	"1"	"0"	"1"	"0"	"1"	"0"	"1"	"0"	"1"	"0"	"1"	"0"	"1"	"0"

#### 3.2.5.2 Time Code Data Format

The time code data is collected from the spacecraft for transmission starting at the minor frame boundary immediately following each Line Sync Code. The time code data is received from the spacecraft and inserted into six (6) contiguous minor frames. Each of the 16 groups of 5 data words within a given minor frame will contain a single bit binary value (0 or 1) of information that is replicated for all of the bits in the group (40 bits per group). Table 2 presents the position of each information bit for the six minor frames. The Time Code data is 480 data words and conforms to the Minor Frame structure as shown in Figure 13, and pre-empts all minor frame video except Band 6 data. The Time Code information is encoded in "8421" (natural) Binary-Coded Decimal (BCD) except for 0.0625 msec which is binary. Transmission order is left to right, top to bottom, odd numbered groups first, then even numbered groups last.



Mf 0, 2, 4 . . . – odd data words Mf 1, 3, 5 . . . – even data words

Representation: 16 Groups of 5 Words Plus 1 Group for Band 6 Plus Spare Word

Figure 13. Minor Frame Data Structure

TABLE 2. TIME CODE DATA FORMAT (6 MINOR FRAMES)

MF#	1A-E	2A-E	3A-E	4A-E	5A-E	6A-E	7A-E	8A-E	9A-E	10A-E	11A-E	12A-E	13A-E	14A-E	15A-E	16A-E
1	0	1	0	1	0	1	0	1	0	1	0	1	0	1	0	1
2	0	0	10 Days (8)	1 Day (8)	0	1 Hr (8)	0	1 Min (8)	0	1 Sec (8)	100 msec (8)	10 msec (8)	1 msec (8)	1/16 msec (8)	SCID #1 (MSB) '0'	1
3	0	0	10 Days (4)	1 Day (4)	0	1 Hr (4)	10 Min (4)	1 Min (4)	10 Sec (4)	1 Sec (4)	100 msec (4)	10 msec (4)	1 msec (4)	1/16 msec (4)	SCID #2 '1'	1
4	0	100 Days (2)	10 Days ( <u>2</u> )	1 Day ( <u>2</u> )	10 Hrs (2)	1 Hr (2)	10 Min (2)	1 Min (2)	10 Sec (2)	1 Sec (2)	100 msec (2)	10 msec (2)	1 msec (2)	1/16 msec (2)	SCID #3 '1'	1
5	0	100 Days (1)	10 Days (1)	1 Day (1)	10 Hrs (1)	1 Hr (1)	10 Min (1)	1 Min (1)	10 Sec (1)	1 Sec (1)	100 msec (1)	10 msec (1)	1 msec (1)	1 msec (1)	SCID #4 (LSB) '1'	1
6	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0

(n) = BCD Weight (8, 4, 2, or 1) except column 15A-E (natural binary)

msec = milliseconds

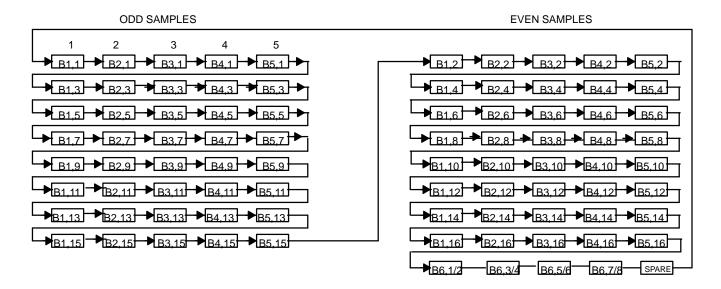
[n] = Binary coded fraction weight

#### 3.2.5.3 Scene Data Formats

Scene data is provided in two specified Scene Data formats (Scene Data Format 1 and Scene Data Format 2). Scene data transmission starts at the minor frame boundary immediately following the Time Code and conforms to the Minor Frame Data Structure. Transmission of scene data continues until the start of the next End of Line Pattern code. For reference, 6313 Minor Frames of scene data are nominally transmitted during any given scan cycle. The digitized scene data from the analog video inputs can be organized into either of two minor frame scene data formats. A given multiplexer is capable of simultaneously providing both formats and has two high rate serial digital outputs that are each allocated to providing one of the two formats. Table 3 lists the spectral bands allocated to each of the two formats. Figure 14 presents the scene data pixel pattern for Scene Data Format 1 which includes bands 1 through 6. Figure 15 presents the specific scene data pixel pattern for Scene Data Format 2 which includes bands 6, 7, and PAN.

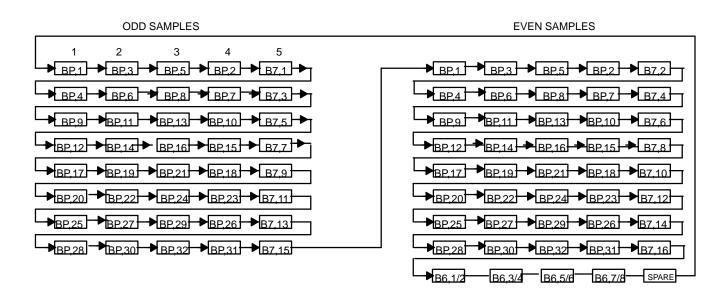
#### 3.2.5.4 End of Line Pattern Code Format

The occurrence of the line stop pulse generated by the ETM+ scan mirror assembly is asynchronously detected by the L–7 AEM. Upon detection of the line stop pulse, an End of Line Pattern Code is generated and is synchronized to the Minor Frame boundary of the next minor frame. The End of Line Pattern code conforms to the Minor Frame Data Structure and preempts all minor frame video except Band 6 data. The length of the End of Line Pattern Code included in 2 consecutive minor frames is 160 bytes. The End of Line Pattern Code will be inserted into two contiguous minor frames. Each of the 16 groups of 5 data words within the minor frame will contain a single bit binary value (0 or 1) that is replicated for all of the bits in the group (40 bits per group). Table 4 lists the single bit values to be used. The values listed in Table 4 are used for both Minor Frames.



Representation: BX,Y = Band X, Detector Y; Band 6 = Primary

Figure 14. Scene Data Format 1



Representation: BX,Y = Band X, Detector Y; Band 6 = Redundant

Figure 15. Scene Data Format 2

TABLE 3. SCENE DATA FORMATS

BAND	INPUT	SAMPLE RATE	GROUND FOV	FORMAT 1	FORMAT 2
1	16	104 kHz	30 meters	Х	
2	16	104 kHz	30 meters	Х	
3	16	104 kHz	30 meters	Х	
4	16	104 kHz	30 meters	Х	
5	16	104 kHz	30 meters	Х	
6	8	52 kHz	60 meters	Х	х
7	16	104 kHz	30 meters		х
PAN	32	208 kHz	15 meters		Х

#### 3.2.5.5 Scan Line Length/Direction Data

The Scan Mirror Assembly (SMA) may be operated in either of two modes: (a) the primary mode, referred to as the Scan Angle Monitor (SAM) Mode, and (b) the back-up, or Bumper Mode. The primary SAM mode uses feedback from three split optical SAM sensors (at both ends and the middle of the mirror travel) to correct the motion of the scan mirror to keep the first half and second half scans as identical as possible, by applying two torque pulses during each turn-around (the timing constants of the pulses being adjusted based on the behavior of the scan mirror during the previous forward and reverse scan). In contrast, the back-up Bumper mode also applies two torque pulses during the turn-arounds, but their time constants are fixed (chosen so that the SAM and Bumper operational characteristics shall be as similar as practical) and simulated SAM pulses are issued to maintain synchronization of the ETM+ shutter, scan line corrector, the SMA processor and the wide band data.

In each mode, scan line length/direction data is collected from the SMA for transmission starting at the Minor Frame boundary immediately following each End of Line Code. The scan line length/direction data conforms to the minor frame data structure and preempts all minor frame video data except band 6 data. The length of the scan line data is 160 data words (2 minor frames). Scan line data is received from the SMA with the ETM+ scanner and inserted into two (2) contiguous minor frames. Each of the 16 groups of 5 data words within a given minor frame will contain a single bit binary value (0 or 1) of information that is replicated for all bits in the group (40 bits per group). All scan line information corresponds to the scan prior to the current scan.

The position of each information bit for two minor frames is described in Table 5A for the primary, SAM mode and it Table 5B for the back-up, Bumper mode. In the SAM mode, the scan length information is given in two (2) twelve (12) group fields and the scan direction in an eight (8) group field. In the telemetry stream, the three (3) fields are interleaved as shown in Table 5A; together, they total 32 groups of 5 data words and comprise the two minor frames of scan line length/direction data. The first field of data "SHSERR" is the second half scan error (in counts) from a nominal 161,165. The second data field "FHSERR" is the first half scan error (in counts) from a nominal 161,164. Both scan error fields are comprised of a sign bit and 11 binary weighted bits, most significant bit first. Negative magnitudes (sign = 1) are two's complement. The active scan time (in seconds) can be computed using the following formula:

Active Scan Time 
$$(T_{AST}) = 2 \times \left\{ (161,164 - T_{FHSERR}) + (161,165 - T_{SHSERR}) \right\} \times \left[ \frac{120}{119} \times \left( \frac{7}{74.914 \times 10^6} \right) \right]$$

The third data field gives the direction of the scan. A forward scan (West to East, descending node) is denoted by ones. A reverse scan is denoted by all zeros.

In the case of the Bumper mode, the two minor frames contain two interleaved fields, as shown in Table 5B. The first field of data, a twenty–four (24) group field, is the Bumper-to-Bumper (B–B) time, which is the absoluter transition time in counts from bumper to bumper. The most significant bit is again first. The nominal Bumper-to-Bumper time is 71.820 msec and can be computed (in seconds) as follows:

Bumper-to-Bumper Time (T<sub>BTBT</sub>) = 
$$2 \times T$$
 (B–B counts)  $\times \left[ \frac{120}{119} \times \left( \frac{7}{74.914 \times 10^6} \right) \right]$ 

The second data field in the Bumper mode is the same as the third field in the SAM mode and again gives the scan direction: A forward scan (West to East, descending node) is denoted by ones. A reverse scan is denoted by all zeros.

TABLE 4. END OF LINE PATTERN CODE (1 OF 2 MINOR FRAMES)

I	1A-E	2A-E	3A-E	4A-E	5A-E	6A-E	7A-E	8A-E	9A-E	10A-E	11A-E	12A-E	13A-E	14A-E	15A-E	16A-E
ľ	"0"	"0"	"0"	"0"	"0"	"0"	"0"	"0"	"1"	"1"	"1"	"1"	"1"	"1"	"1"	"1"

TABLE 5A. SAM MODE SCAN LINE DATA (2 MINOR FRAMES)

		GROUPS OF 5 DATA WORDS														
MF#	1–AE	2A-E	3А-Е	4A-E	5A-E	6A-E	7A-E	8A-E	9A-E	10A-E	11A-E	12A-E	13A-E	14A-E	15A-E	16A-E
1	SHS ERR Bit 1 (±)	SHS ERR Bit 9	SHS ERR Bit 2 (MSB)	SHS ERR Bit 10	SHS ERR Bit 3	SHS ERR Bit 11	SHS ERR Bit 4	SHS ERR Bit 12 (LSB)	SHS ERR Bit 5 (LSB)	FHS ERR Bit 1 (±)	SHS ERR Bit 6	FHS ERR Bit 2 (MSB)	SHS ERR Bit 7	FHS ERR Bit 3	SHS ERR Bit 8	FHS ERR Bit 4
2	FHS ERR Bit 5	SCN DIR	FHS ERR Bit 6	SCN Dir	FHS ERR Bit 7	SCN DIR	FHS ERR Bit 8	SCN DIR	FHS ERR Bit 9	SCN DIR	FHS ERR Bit 10	SCN DIR	FHS ERR Bit 11	SCN DIR	FHS ERR Bit 12 (LSB)	SCN DIR

SHS ERR: Second Half Scan Error (± counts from nominal value of 161,165 counts).

FHS ERR: First Half Scan Error (± counts from nominal value of 161,164 counts).

SCN DIR: Scan Direction ("0" = Reverse, "1" = Forward).

TABLE 5B. BUMPER MODE SCAN LINE DATA (2 MINOR FRAMES)

		GROUPS OF 5 DATA WORDS														
MF#	1-AE	2A-E	3A-E	4A-E	5A-E	6A-E	7A-E	8A-E	9A-E	10A-E	11A-E	12A-E	13A-E	14A-E	15A-E	16A-E
1	B-B Time Bit 1 (MSB)	B–B Time Bit 9	B–B Time Bit 2	B–B Time Bit 10	B–B Time Bit 3	B–B Time Bit 11	B–B Time Bit 4	B–B Time Bit 12	B–B Time Bit 5	B–B Time Bit 13	B–B Time Bit 6	B–B Time Bit 14	B–B Time Bit 7	B–B Time Bit 15	B–B Time Bit 8	B–B Time Bit 16
2	B–B Time Bit 17	SCN DIR	B-B Time Bit 18	SCN DIR	B–B Time Bit 19	SCN DIR	B–B Time Bit 20	SCN DIR	B–B Time Bit 21	SCN DIR	B–B Time Bit 22	SCN DIR	B–B Time Bit 23	SCN DIR	B-B Time Bit 24 (LSB)	SCN DIR

B–B Time: Absolute transition time from bumper to bumper in counts.

SCN DIR: Scan Direction ("0" = Reverse, "1" = Forward)

#### 3.2.5.6 Calibration Data

#### 3.2.5.6.1 Internal Calibration System

An internal calibration system is provided within the ETM+ instrument to assists in performing radiometric calibration of image data. This system consists of an obscuration shutter assembly which includes a set of calibration source lamps with associated optical conductors for bands 1 through 5, 7, and Panchromatic band. A temperature controlled blackbody surface is employed as a calibration source for the thermal band (Band 6).

The obscuration shutter also provides a dark surface for use in setting the DC reference level for Bands 1 through 5, 7, and Pan with the instrument electronics, and provides a second known temperature surface for use in the thermal band calibration.

Calibration and DC reference recognition is achieved at the completion of each imaging scan with the obscuration shutter rotating into view of the detectors. As the shutter passes through these optical paths, the calibration lamps, blackbody surface, and the dark surfaces pass through the detector's field of view. As a result, a calibration radiance signal and a DC restore level are provided between active scans for bands 1 through 5, 7, and Pan detectors. For the Band 6 detectors, temperature levels are provided between active scans. These data are intended for use in establishing system calibration criteria during image processing.

During each mirror turnaround period, the internal calibration system for Bands 1 through 5, 7, Band 6, and the Pan Band provides calibration data over approximately 143 minor frames (1374 µsec) preceding the reverse scan and 153 minor frames (1470 µsec) preceding the forward scan. Refer to Table 7 and Figures 16 and 17 for further details regarding nominal start and stop minor frame locations of the Line Sync Code, Time Code, End-of-Line (EOL), Scan Line Length and Scan Direction, Shutter Obscuration, the internal Calibration Period, DC Restore and Postamble Data. Note that the start and stop minor frame locations are approximate and may vary with Scan Mirror Assembly operating conditions and over time.

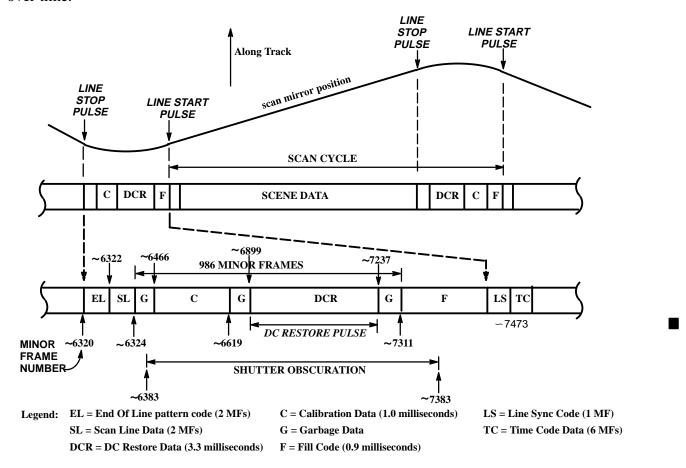


Figure 16. Major Frame Events (Reverse Scan)

Two lamps are available to provide calibration stimuli for Bands 1 through 5, 7, and the Pan Band. The circuitry for each lamp drive is unique such that the calibration lamp circuitry and lamp are completely redundant. Each lamp has two states (On or Off). Calibration data format is just like that of scene data.

#### 3.2.5.6.2 Multiplexer Bi-State Gains and A/D Conversion

Each spectral band may be uniquely commanded to either of two gain states (High or Low). In addition, each multiplexer has dual redundant circuitry for Band 6 and the gain for each circuit may be set independently. The high gain state increases the instrument's capability to detect subtle signal changes, while the low gain feature is intended for scenes of high reflectivity and greater scene dynamic range. This operational capability is intended to be used according to high or low radiance of geographic areas.

For reference, Table 6 presents the Band to A/D Converter assignment summary.

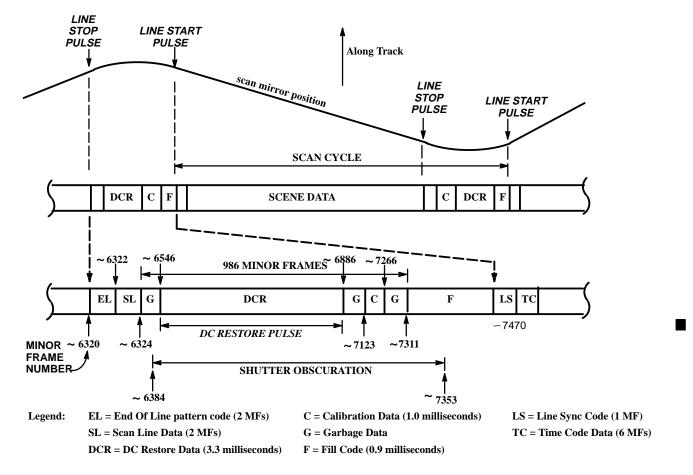


Figure 17. Major Frame Events (Forward Scan)

#### 3.2.5.6.3 DC Restore System (Band 1 through 5, 7 and Pan)

DC restore is a technique for minimizing the effects of low frequency noise and drift. A dark level is applied to the sensors during shutter obscuration to develop a zero clamp level for the analog-to-digital circuitry. This zero clamp level is fractionally updated before each scan to a nominal level of 10 digital counts in the low gain state and 20 digital counts in the high gain state. The zero-clamp level observed during the shutter-closed period should be considered representative of a sensor black-level output during imaging.

The dark condition is established when the shutter blocks the normal optical path for each detector, and is present throughout the obscuration period except during the calibration pulse. Shutter obscuration occurs approximately 0.62 milliseconds after EOL on both forward and reverse scans, and continues for approximately 9.42 milliseconds. Adjustment of the DC restore level lasts approximately 3.25 milliseconds.

#### 3.2.5.6.4 Thermal Band Radiometric Calibration System

A temperature controlled blackbody and a temperature-measured shutter surface provide the calibration reference points for the eight Band 6 detectors. Band 6 detectors view the temperature-measured shutter surface during the DC restore calibration period of each mirror scan. The calibration shutter and blackbody temperatures are measured and inserted into each PCD minor frame (Subcommutation word 72, minor frames 76 and 78, respectively). Absolute calibration will be necessary for the Thermal

Infrared band to account for the blackbody shading factor. Compensation for temperature drift and possible emmissivity variations may be required throughout the mission.

#### 3.2.5.7 Fill Code Format

Generation of Fill Code begins at 986 minor frames following the transmission of the Scan Line data. The fill code conforms to the Minor Frame Data structure and preempts all minor frame video except Band 6 data. The transmission of fill code continues until the start of the next data word after the next line start pulse is detected (approximately 161 minor frames) in vacuum and 179 minor frames in air. Fill Code will be generated and inserted into each successive minor frame until the occurrence of the next scan line start pulse. The Fill Code will terminate on byte or word boundaries. The number of Fill Code data words will vary with variations in the Scan Line Length. Each of the 16 groups of 5 data words within a given minor frame will contain a single bit binary value (0) that is replicated for all of the bits in the group (40 bits per group). Table 8 lists the single bit values.

#### 3.2.6 PCD/STATUS DATA

For each transfer frame, ten PCD/STATUS' data words shall be provided to accommodate the insertion of payload correction data (PCD) and ETM+ status data into the VCDU Data Zone shown in Figure 12. Table 9 shows the format and the data content of the PCD/STATUS' data field. The "msb" is transmitted first. The PCD/Status data, data pointer, data pointer BCH error control, time code data, and scan line data inserted in formats 1 and 2 are identical.

PCD data words contain telemetry information regarding the status of the spacecraft and its subsystems.

Status data words shown in Table 9 reflect the current operational status of Landsat 7 subsystems at the start of the first full minor frame within the VCDU Data Zone. The Status data words shall include a minor frame counter, scan direction, and system status. The minor frame counter represents the minor frame count value (range 0 to 8191 (1FFF<sub>16</sub>)) of the minor frame preceding the first full minor frame of the CADU; that is, a running minor frame counter value for every CADU completed. However, for the CADU in which scan line start occurs, both the minor frame counter and the data pointer (PNTR) may not be valid. The minor frame counter is reset upon detection of scan line start (SLS) and the pointer is resynchronized at the start of the first full minor frame in the subsequent CADU.

#### Data PNTR

The Data PNTR follows a defined sequence between scans. The next PNTR may be calculated per equation below:

#### Equation 1: PNTR Sequence

If Pntr <47

New Pntr = Current Pntr -47

If Pntr  $\geq 47$ 

New Pntr = Current Pntr +38

If Scan Line Start is detected within the CADU the current PNTR is resynchronized and will not be guaranteed to be valid. The next valid PNTR is based on the data contained in the next consecutive CADU.

Note: All numbers are in decimal.

MF Counter

The MF differential count is related to the data PNTR and may be calculated by the equation shown below. Minor Frame Count (PNTR) – Minor Frame Count (PNTR–1) = Minor Frame Differential Count (PNTR-1).

Equation 2: Complete MF\_Diff\_Count

If Pntr  $\geq 48$ 

MF differential count number of complete Minor frames = 11.

If Pntr < 48 and If Pntr  $\neq$  0

MF differential count number of complete Minor frames = 12.

If Pntr = 0

MF differential count number of complete Minor frames = 11.

Note: All numbers are in decimal.

If a Scan Line Start is detected within the CADU the current Minor Frame counter will reset based on the location of scan line start within the CADU. The new counter is based on data contained within the next consecutive CADU.

#### 3.2.7 PCD FORMATS

Payload Correction Data (PCD) contains all data required by ground stations to geometrically correct ETM+ sensor data and redundantly provides the ETM+ imaging configuration. The PCD data is embedded in every wideband data VCDU at a rate of four bytes of PCD per VCDU as shown in Figure 12. The types of PCD data and formatting are provided in this section.

TABLE 6. BAND TO A/D CONVERTER ASSIGNMENTS

ANALOG CHANNELS	BAND	A/D 1	A/D 2	A/D 3	A/D 4	A/D 5	A/D 6
Format 1							
	Band 1	X <sub>A</sub>					
	Band 2		X <sub>A</sub>				
	Band 3			X <sub>A</sub>			
	Band 4				X <sub>A</sub>		
	Band 5					X <sub>A</sub>	
	Band 6 <sub>P</sub>						X <sub>B</sub>
Format 2							
	Pan	X <sub>B</sub>	X <sub>B</sub>	X <sub>B</sub>	X <sub>B</sub>		
	Band 6 <sub>R</sub>					X <sub>B</sub>	
	Band 7						X <sub>A</sub>

Legend:

P = Primary

A = A side of card

P = Primary A = A side of card R = Redundant B = B side of card

#### 3.2.7.1 Unpacked PCD Format

The PCD data, which is asynchronous with the ETM+, is generated at 4 kbytes/sec. The unpacked PCD format is the PDF data passed to the ETM+ which is shown in Figure 18.

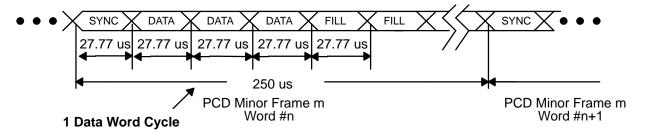


Figure 18. Unpacked PCD Format and Timeline

The PDF utilizes a state machine to perform the transfer of PCD data to the payload. The state machine must be incremented through the five (5) state sequence shown in Figure 18. The sequence, as shown, consists of a Sync word, one valid PCD data word repeated three (3) times, and a Fill word. The fill word remains in the PDF output buffer until the next data word cycle starts, so any further attempt by the payload to read PCD data simply results in another fill word transfer. As a result, a minimum of 8 words are generated during each PCD data word cycle.

#### 3.2.7.1.1 Unpacked PCD Definitions

The unpacked PCD data may consist of three types of eight bit data words: Sync words, data words or Fill data words. The values for these data words are as follows:

1. Sync: 16<sub>16</sub>

2. Data:  $00_{16} \leftrightarrow FF_{16}$ 

3. Fill: 32<sub>16</sub>

#### 3.2.7.2 Packed PCD Format

The packed PCD format is constructed by ground software, by finding the Sync word in the unpacked data stream, extracting the data words, performing a bit-wise majority rule of the three consecutive data words to select one of the three words, and then storing the selected data into a buffer for interpretation. In the packed format, complete PCD cycle requires a period of 16.384 seconds. Each PCD cycle is composed of four PCD major frames with each major frame having a period of 4.096 seconds. Cycle refers to a complete set of a PCD table of data.

TABLE 7. DATA FORMAT

DATA STREAM EVENT	EVENT DESCRIPTION	FORWARD SCAN START MF	FORWARD SCAN STOP MF	REVERSE SCAN START MF	REVERSE SCAN STOP MF
Line Sync Code	MF Sync Code preempts all video stream data except Band 6, 5 '1's, 5 '0's (eight times)	0	0	0	0
Time Code	6 Ms of time code received from S/C preempts all video stream data except Band 6	1	6	1	6
End-of-line	Two MF's, 40 '0's, 40 '1's, 40 '0's, 40 '1's	~6320 ±1 MF	~6321 ±1 MF	~6320 ±1 MF	~6321"1 MF
Scan line length & Scan Direction	2 MFs for the scan just prior to the current scan, preempts all video except band 6	~6322 ±1 MF	~6323 ±1 MF	~6322 ±1 MF	~6323"1 MF
Shutter Obscuration	Calibration Shutter (~960 MF) Calibration Shutter (Pan Only) Calibration Shutter (Band 4 Only) Calibration Shutter (Band 5 Only) Backup Shutter (~780 MF)	~6384 ~6390 ~6384 ~6385 ~6476	~7353 ~7353 ~7325 ~7306 ~7256	~6383 ~6383 ~6401 ~6418 ~6551	~7383 ~7383 ~7375 ~7383 ~7331
Internal Calibration period	Bands 1–5, 7, Pan Pan Only Band 1 Only Band 2 Only Band 3 Only Band 4 Only Band 7 Only Band 5 Only Band 6 Blackbody calibration period	~7123 ~7147 ~7143 ~7138 ~7137 ~7135 ~7125 ~7123 ~7123	~7266 ~7266 ~7258 ~7257 ~7248 ~7243 ~7233 ~7229 ~7229	~6466 ~6466 ~6475 ~6477 ~6483 ~6490 ~6498 ~6501 ~6488	~6619 ~6595 ~6597 ~6605 ~6607 ~6610 ~6618 ~6619 ~6612
DC Restore	Calibration shutter Backup Shutter	~6546 ~6696	~6886 ~7039	~6899 ~6783	~7237 ~7126
Postamble Data: Vacuum Operation	161 MFs of fill data preempts all Video data except BD 6	~7311 ±1 MF	~7470 ±1 MF	~7311 ±1 MF	~7473 ±1 MF
Ambient Operation	179 MFs of fill data preempts all Video data except BD 6	~7311 ±1 MF	~7488 ±1 MF	~7311 ±1 MF	~7491 ±1 MF

# TABLE 8. FILL CODE (1 MINOR FRAME)

I	1A–E	2A-E	ЗА-Е	4A-E	5A-E	6A-E	7A-E	8A-E	9A-E	10A-E	11A-E	12A-E	13A-E	14A-E	15A-E	16A-E
Γ	"0"	"0"	"0"	"0"	"0"	"0"	"0"	"0"	"0"	"0"	"0"	"0"	"0"	"0"	"0"	"0"

TABLE 9. PCD/STATUS DATA

WORD#	DATA WORD INFO	DATA WORD FORMAT				
1	PCD 1	Bit 1 thru 8 = 1st PCD word (bit 1 = msb)				
2	PCD 2	Bit 1 thru 8 = 2nd PCD word (bit 1 = msb)				
3	PCD 3	Bit 1 thru 8 = 3rd PCD word (bit 1 = msb)				
4	PCD 4	Bit 1 thru 8 = 4th PCD word (bit 1 = msb)				
5	MF count (hi)	Bit 1 = scan direction; "0" = Reverse (East to West), "1" = Forward (West to East) Bits 2 and 3 = spare; each set to "0" Bits 4 thru 8 = most significant 5 bits of minor frame count within given scan				
6	MF count (low)	Bits 1 thru 8 = least significant 8 bits of minor frame count within given scan				
7	System status  Bits 1 thru 3 = MUX Assembly ID; multiplexer assemblies 0 thru 7 identified  Bit 4 = Format 1/2 ID; "0" = FMT 1, "1" = FMT 2  Bit 5 = Spare; set to "0"  Bit 6 = Shtr 1/2; "0" = CAL Shtr, "1" = Backup Shtr  Bit 7 = Spare; set to "0"  Bit 8 = PAN band gain; "0" = LO, "1" = HI					
8	Band gains	Bits 1 thru 8 correspond respectively to Bands 1 thru 5, Format 1 Band 6, Format 2 Band 6, and Band 7. For each bit, low gain = "0" and high gain = "1".				
9	Spare	Bits 1 thru 8 each set to "0" (fill data)				
10	10 Spare Bits 1 thru 8 each set to "0" (fill data)					
Note: The	Note: The msb is transmitted first					

#### **3.2.7.2.1** *PCD Major Frame*

A PCD major frame consists of 128 PCD minor frames. Each PCD minor frame consists of 128 PCD words. Four PCD major frames are necessary for a PCD Cycle. Figure 19 shows the PCD major frame format. Transmission order is top to bottom, left to right in the major frame format.

#### 3.2.7.3 PCD Minor Frame

The PCD minor frame consists of 128 eight bit words as shown in Table 10–1. The majority of these 128 words consist of ADS and Gyro data. Also included are the sync word, minor frame ID MFID, major frame ID and a subcomm word (word 72) as shown in Figure 19. The subcomm words include attitude, gyro drift, ephemeris, time code, ADS Temperature, PCD multiplexer status.

#### 3.2.7.3.1 PCD Sync

The PCD sync word identifies the start of the minor frame and it is defined in Table 10–2.

#### 3.2.7.3.2 PCD Minor Frame ID Counter

The PCD minor frame ID counter appears in every minor frame in word location 65. The range of the counter is 0 to 127 and is shown in Table 10–3.

#### 3.2.7.3.3 PCD Major Frame ID

The PCD Major Frame ID appears in the subcom word in the second, third and fourth major frames and further described in 3.2.7.4.14.

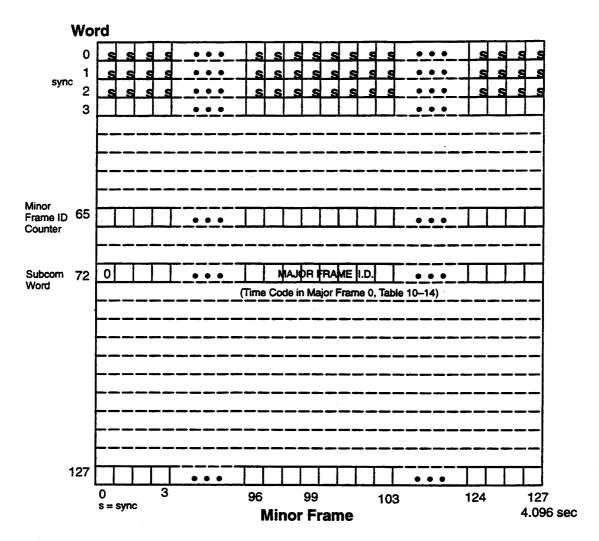


Figure 19. ETM+ PCD Major Frame Format

#### 3.2.7.4 PCD Data Types

## 3.2.7.4.1 Angular Displacement Sensor

The Angular Displacement Assembly (ADA) consists of three nominally orthogonal Angular Displacement Sensors (ADS). Each ADS of the ADA is sampled every 2 milliseconds during the odd numbered word time preceding the first of the two data words. The sample is converted to a 12-bit integer value then inserted into two (2) consecutive words of a PCD minor frame (refer to Table 10) with the four most significant bits of the first word set to zero. Digital count (0) is the maximum positive angular displacement and digital count 4095 is the maximum negative angular displacement. The least significant bit of each count is nominally 250/2<sup>11</sup> microradians (or 500/2<sup>12</sup>). The nominal zero angular displacement output of any ADS is 2048 counts. There are 8192 samples of each ADS in a PCD cycle. The most significant bit is transmitted first.

#### 3.2.7.4.2 ADS Temperature

Up to four ADS-related temperatures will be sampled once a PCD major frame (every 4.096 seconds). Each sample will be converted to a 12 bit word and inserted in 2 consecutive words of format, with the 4

msb's of the first word set to zero, as show in Table 11. The data will be sampled in the word time preceding the data word. That is, ADS Temperature #1 is placed in word 72 minor frame 108 and 109 and sampling time is during word 71 of minor frame 108. The formatting and range of ADS temperature data is shown in Table 11.

TABLE 10-1. PCD TYPICAL MINOR FRAME

DATA	MINOR FRAME WORD NUMBER	DATA	MINOR FRAME WORD NUMBER	
Sync	0,1,2	MFID CTR	65	
ADS-X	3,4	ADS-X	66,67	
ADS-Y	5,6	ADS-Y	68,69	
ADS-Z	7,8	ADS-Z	70,71	
0 Filled	9	Sub Comm	72	
0 Filled	10	0 Filled	73	
ADS-X	11,12	ADS-X	74,75	
ADS-Y	13,14	ADS-Y	76,77	
ADS-Z	15,16	ADS-Z	78,79	
Gyro	17	0 Filled	80	
0 Filled	18	Gyro	81	
ADS-X	19,20	ADS-X	82,83	
ADS-Y	21,22	ADS-Y	84,85	
ADS-Z	23,24	ADS-Z	86,87	
0 Filled	25,26	0 Filled	88,89	
ADS-X	27,28	ADS-X	90,91	
ADS-Y	29,30	ADS-Y	92,93	
ADS-Z	31,32	ADS-Z	94,95	
Gyro	33	0 Filled	96	
0 Filled	34	Gyro	97	
ADS-x	35,36	ADS-X	98,99	
ADS-Y	37,38	ADS-Y	100,101	
ADS-Z	39,40	ADS-Z	102,103	
0 Filled	41,42	0 Filled	104,105	
ADS-X	43,44	ADS-X	106,107	
ADS-Y	45,46	ADS-Y	108,109	
ADS-Z	47,48	ADS-Z	110,111	
Gyro	49	0 Filled	112	
0 Filled	50	Gyro 113		

TABLE 10-1. PCD TYPICAL MINOR FRAME (CONT)

DATA	MINOR FRAME WORD NUMBER	DATA	MINOR FRAME WORD NUMBER
ADS-X	51,52	ADS-X	114,115
ADS-Y	53,54	ADS-Y	116,117
ADS-Z	55,56	ADS-Z	118,119
0 Filled	57,58	0 Filled	120,121
ADS-X	59,60	ADS-X	122,123
ADS-Y	61,62	ADS-Y	124,125
ADS-Z	63,64	ADS-Z	126,127

TABLE 10-2. PCD MINOR FRAME SYNC FORMAT

WORD	BCD CODE	HEX CODE
0	(msb) 1111 1010 (lsb)	FA
1	1111 0011	F3
2	0010 0000	20

TABLE 10-3. PCD MINOR FRAME IDENTIFIER

DATA WORD	BIT (msb) 0 1 2 3 4 5 6 7 (lsb)				
65	I <sub>0</sub> , I <sub>1</sub> , I <sub>2</sub> , I <sub>3</sub> , I <sub>4</sub> , I <sub>5</sub> , I <sub>6</sub> , I <sub>7</sub>				

 $I_0 = 0$  at all times.

#### 3.2.7.4.3 Gyro Data

The inertial measurement unit (IMU) consists of three, two-degree-of-freedom gyros. The spacecraft maintains 24 bit pulse accumulators for each gyro axes (XA, XB, YA, YB, ZA, ZB). The gyro pulse counts from each of three selected accumulators are sampled every 64 milliseconds. The spacecraft provides one data value for each IMU axis in the PCD. The IMU generates a signed pulse for each 0.061 arc-sec of angular motion. A positive pulse increments a 24-bit register and a negative pulse decrements the register. A positive pulse is generated by a negative rotation about the gyro axis. The pitch orbital motion and gyro drift cause the register to periodically overflow. The register is reset to zero when its value is positive  $2^{23}$ -1 and a positive pulse is received, or when its value is negative  $2^{23}$  and a negative pulse is received. Each axis is sampled at the same time. There are 256 samples of each IMU axis during a PCD cycle. The sample timing is as follows:

Let the gyro samples for any one axis appearing in a PCD cycle (16.384 seconds) be numbered N = 0, 1, ..., 255. Then the time for each sample is the PCD time code plus (64N) milliseconds.

 $I_1 - I_7 = Binary count from 0_{10} to 127_{10} (msb - lsb)$ 

TARIF 11	ADS/PDF TEMPERATI	DEC
IADI/P/II.	AUS/PUR IRWIPERALL	KIL

		WORD 72 MINOR FRAMES	DATA WORD	SAMPLED DURING WORD
Temp 1	(ADS-X)	108 109	(msb) 0000XXXX XXXXXXXX (lsb)	71 (108)
Temp 2	(ADS-Y)	110 111	(msb) 0000XXXX XXXXXXXX (lsb)	71 (110)
Temp 3	(ADS-Z)	112 113	(msb) 0000XXXX XXXXXXXX (lsb)	71 (112)
Temp 4	PDF Analog Temperature	114 115	(msb) 0000XXXX XXXXXXXX (lsb)	71 (114)

Applies to the four temperature sensors:

Range =  $+50^{\circ}$ C =  $0000_{16}$  to  $0^{\circ}$ C = OFFF<sub>16</sub>.

lsb =  $0.0122^{\circ}1C$  (4096/50)

Each data value consists of a 23-bit value which is provided in the PCD as three, eight-bit words. The data are in two's complement format with the msb first. Figure 20 shows the format of the gyro data and its position in the PCD. Note, the gyro data is in the IMU reference frame. The relationship between the IMU reference frame and the satellite Navigational Reference Frame for each gyro axis is defined in the Landsat 7 Program Coordinate System Standard, 23007610.

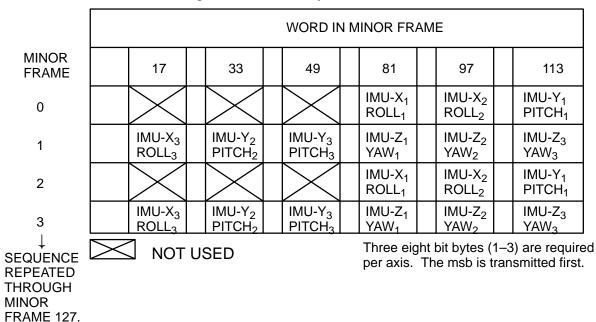


Figure 20. PCD Gyro Data

#### 3.2.7.4.4 Gyro Drift Data

The gyro drift calculation is performed based upon asynchronous star sightings and are presented in the PCD subcomm word 72 for PCD major frame (0) and calculated at the PCD cycle time code (TC) minus 8.192 seconds. Gyro drift is calculated in the nagivation axis coordinate system. The Units of gyro drift rate are radians/512 msec and the data are calibrated at an lsb weight of 2<sup>-47</sup>. The format and frame position of the gyro drift binary scaled integer data is shown in Figure 21. Accurate data are available when PRADS is converged. If PRADS is both not running nor converged gyro drift data are not valid.

Word 72	msb 2 <sup>-17</sup>				Isb 2 <sup>-47</sup>
	sxxxxxx	xx	xxxxxx	xxxxxxx	xxxxxxx
THE	TA BX	16	17	18	19
THE	TA BY	20	21	22	23
THE	TA BZ	24	25	26	27

Figure 21. Gyro Drift Data

The gyro drift data will appear in word 72 of minor frames 16 through 27 of PCD major frame zero.

#### 3.2.7.4.5 Attitude Estimate

The spacecraft calculates an estimate of the attitude, represented as Euler parameters. The Euler Parameters are labeled EPA1, EPA2, EPA3, and EPA4 and are components of the quaternion defining the rotation. Components 1 through 3 define the eigen-axis of rotation in Earth Centered Inertial (ECI) coordinates, and component 4 defines the rotation about that axis. Euler double precision words are scaled to 32 bits in two's complement form as shown in Figure 22. Accurate data are available when PRADS is running and converged. If PRADS is not running, backup (YGC) data will be provided.

	msb 2 <sup>0</sup>				Isb 2 <sup>-30</sup>	
Bit Order (msb)	SXXXXXXX	xxxxxxx	XX	xxxxx	XXXXXXXX	(Isb)
Byte Order (MSB)	1	2		3	4	(LSB)
Content	PCE	Minor Frame	e Numbe	ers		
EPA1	0	1	2	3		
EPA2	4	5	6	7		
EPA3	8	9	10	11		
EPA4	12	13	14	15		
Byte Order	1	2	3	4		

Figure 22. Attitude Estimate

Four Euler Parameters (EPA's) are output in word 72 of minor frames 0 through 15 of each PCD Major Frame. The time associated with attitude data contained within the PCD can be derived from the time code contained in words 96 through 102 of the first PCD major frame in the cycle. The attitude data time is derived as follows:

The output sequence is shown in Figure 22. The most significant bit is transmitted first.

#### 3.2.7.4.6 Time of Last (SV) Space Vehicle Clock Update

The time of the last SV clock update is inserted in the PCD stream. The SV clock is typically updated by the MOC once per day, and during ETM+ non-imaging periods. The bit format is a 48-bit extended precision floating point value in seconds from midnight of the first day of the year as shown in Figure 23 (See MIL–STD–1750A). The 48 bits of last SV clock update are subcommutated into word 72 of Minor Frames 28 thru 33 of the first PCD major frame in the PCD cycle.

_	PCD Major Frame Number	Time Computation			
	0 1 2 3	PCD time code – 8.192 seconds PCD time code – 4.096 seconds PCD time code + 0.000 seconds PCD time code + 4.096 seconds	S S		
MF No 28 29 30 31 32 33	First 8 bits of the time of the last SV clock parameter upload (msb) Second 8 bits of the time of the last SV clock parameter upload Third 8 bits of the time of the last SV clock parameter upload Fourth 8 bits of the time of the last SV clock parameter upload Fifth 8 bits of the time of the last SV clock parameter upload Sixth 8 bits of the time of the last SV clock parameter upload (lsb)				
	0 1 (msb)	23 24	31 32	47 (Isb)	

Figure 23. SCP Representation of a 48-bit Extended Floating Point Word

8-bit exponent

Last 16-bits of mantissa

#### 3.2.7.4.7 SV Time Drift Characterization Data

Sign

The S/C Time Drift Characterization Data is used on the ground, along with the Time of Last S/C Clock Update, to correct the spacecraft time, reported in the PCD and video, for clock drift, to within  $\pm 15$  milliseconds of UTC. The SV time drift characterization data is updated by the MOC daily, and at the same time as the SV clock update data during ETM+\_non-imaging periods. The corrected time is calculated according to the following equation:

$$\Delta t = t_s/c - t_{update}$$
  
 $T_C = t_s/c + C_0 + C_1 \Delta t + 0.5 C_2 \Delta t^2$ 

First 23 bits of mantissa

where:

 $\Delta t$  = the spacecraft clock time relative to the last clock update

 $T_{\rm C}$  = the spacecraft corrected time (approx. UTC,  $\pm 15$  milliseconds)

ts/c = a spacecraft clock time; the clock time for ephemeris calculations is the time of 2 Hz cycle that the spacecraft state is referenced to

tupdate = the spacecraft clock time of the last ground commanded clock update

C0 = the clock correction bias term – can be used to minimize the clock error over some span of time; may be set to zero if not needed

C1 = the clock correction first order coefficient (drift rate)

C2 = the clock correction second order coefficient (drift acceleration); may be set to zero if not needed

The 32 bits of S/C Time Drift Characterization Data are subcommutated into word 72 of Minor Frames 36–41 of the first PCD Major Frame in the PCD cycle. The msb is transmitted first.

MF No.	DESCRIPTION		
36	First 8 bits of the clock correction bias term (msb)	1	
37	Second 8 bits of the clock correction bias term (lsb)	}	msec.
38	First 8 bits of the clock correction first order coefficient (drift rate) (msb)	1	msec./day
39	Second 8 bits of the clock correction first order coefficient (drift rate) (lsb)	ſ	msec./day
40	First 8 bits of the clock correction second order coefficient (drift accel) (msb)	) [	msec./day <sup>2</sup>
41	Second 8 bits of the clock correction second order coefficient (drift accel)(lsb	•	msec./day

# 3.2.7.4.8 *Ephemeris*

Four ephemeris points are provided by the spacecraft during a PCD cycle and inserted into the PCD stream. The ephemeris entry consists of six components: Position coordinates X, Y, and Z in kilometers and Velocity components X', Y', and Z' in kilometers per second. The coordinate system is the J2000 and is defined in PS23007610 Program Coordinates System Standard (PCSS). These calculations are performed at the same time as the Euler Parameter attitude estimate calculation.

Ephemeris data are 32 bit, two's complement, binary numbers. The range of the Position component is  $\pm -8.3886 \times 10^6$  meters and the range for the Velocity component is  $\pm -8.0$  meters/millisecond. The value of the least significant bit is  $2^{-8}$  meters for position and  $2^{-28}$  meters/millisecond for velocity. The format of this data is shown in Figure 24.

The time associated with ephemeris data contained within the PCD can be derived from the time code contained in words of minor frame 96 through 103 of the first PCD major frame in the PCD cycle. The Ephemeris data time is derived as follows:

Major Frame Number	Time Computation	
0 1 2 3	PCD time code – 8.192 seconds PCD time code – 4.096 seconds PCD time code + 0.000 seconds PCD time code + 4.096 seconds	

The data will appear in word 72 of minor frames 50 through 73 for Major Frames (0) and (2) and minor frames 16 through 39 for Major Frames (1) and (3). The msb is transmitted first.

## 3.2.7.4.9 ETM+ Telemetry Data

DCD

A total of 688 bits of ETM+ telemetry data are inserted into the PCD. The data appears in word 72 of minor frames 16 through 49 of PCD major frame two (2), minor frames 74 though 83 of every major frame and PCD major frame two (2), minor frame word 84. The ETM+ telemetry words which occur in Minor Frames 16 through 49 are sampled approximately once every 16.384 seconds. The ETM+ telemetry words 74 through 80 are sampled by the SCP approximately every 4.096 seconds, then inserted into the next major frame. All other ETM+ telemetry data is sampled approximately every 16.384 seconds including serial word "P" in minor frame word 84 of PCD major frame two (2).

# **Ephemeris in Major Frames (0) and (2)**

Position Compon	ents					
Word 72	MSB 2 <sup>22</sup>				LSB 2 <sup>-8</sup>	
	sxxxxxx	X	xxxxxx	XXXXXXX	XXXXXXX	x
Χ		50	51	52	53	Minor
Υ		54	55	56	57	Frame
Z		58	59	60	61	
Velocity Compo	nents					
Word 72	MSB 2 <sup>2</sup>				LSB 2 <sup>-28</sup>	
	SXXXXXXX	XX	xxxxx	xxxxxxx	xxxxxxx	
X		62	63	64	65	Minor
Υ		66	67	68	69	Frame
Z		70	71	72	73	

# **Ephemeris in Major Frames (1) and (3)**

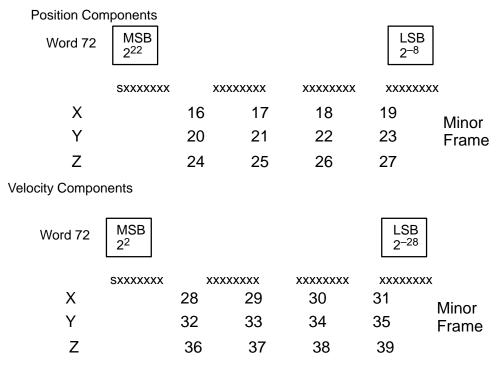


Figure 24. PCD Ephemeris Data

# 3.2.7.4.10 Spacecraft ID and Time Code

Fifty-six bits of spacecraft ID and time code (seven 8-bit words) are inserted in the PCD stream. This code represents the start time for PCD Major Frame (0) and provides the timing reference for all data in the PCD cycle. The 56 bits of spacecraft time code are subcommutated into word 72 of Minor Frames 96 through 102 of PCD Major Frame (0) in the PCD cycle. The output sequence for the 52 time code bits is contained in Table 12. The 4-bit spacecraft ID is (111<sub>2</sub>). The msb is transmitted first. The binary coded decimal (BCD) days field possible range is 0 to 999, hours (0 to 23, minutes (0 to 59), seconds (0 to 59), milliseconds (0 to 999), and binary coded fractional milliseconds with bit values of the fractional milliseconds having values of 1/2, 1/4, 1/8, and 1/16 milliseconds. The day field range is nominally from 001 to 365 or 366 for a leap year and controlled by the the Mission Operations Center.

#### 3.2.7.4.11 Multiplexer Status

The multiplexer gain state is provided in the ETM+ telemetry PCD data. This data is located in Word 72 of Minor frames 82 and 83 of the second Major Frame (Major Frame (1)).

MINOR FRAME NUMBER	TIME CODE WORD NUMBER	WORD 72 BITS 0-7	CONTENT OF WORD 72
96	1	(msb) 0-3 4-7 (lsb)	SPACECRAFT ID HUNDREDS OF DAYS
97	2	0–3 4–7	TENS OF DAYS UNITS OF DAYS
98	3	0-3 4-7	TENS OF HOURS UNITS OF HOURS
99	4	0-3 4-7	TENS OF MINUTES UNITS OF MINUTES
100	5	0–3 4–7	TENS OF SECONDS UNITS OF SECONDS
101	6	0–3 4–7	HUNDRED OF MILLISECONDS TENS OF MILLISECONDS
102	7	0–3 4–7	UNITS OF MILLISECONDS FRACTIONS OF MILLISECONDS (LSB=1/16 MILLISECOND)

TABLE 12. PCD TIME CODE

# 3.2.7.4.12 PDF A/D Ground Reference

The output of the Angular Displacement Assembly (ADA) A/D converter for a grounded input is transmitted in word 72 of minor frames 116 and 117 of each PCD Major Frame. Note that the MSB of minor frame 116 will always be set to "1". An internally generated ground reference signal is sampled and formatted as shown in Table 13. The msb is transmitted first.

TABLE 13. PDF GROUND REFERENCE TELEMETRY

MINOR FRAME	DATA WORD	BIT (msb) 01234567 (lsb)	SAMPLED DURING WORD		
116	72	1000 G <sub>0</sub> G <sub>1</sub> G <sub>2</sub> G <sub>3</sub>	71 (116)		
117	72	$G_4G_5G_6G_7G_8G_9G_{10}G_{11}$	71 (116)		
$G_0$ – $G_{11}$ = Ground	G <sub>0</sub> –G <sub>11</sub> = Ground Reference Bits (msb–lsb)				

# 3.2.7.4.13 Minor Frame Sync

The same sync pattern will appear in words 0 through 2 of each PCD minor Frame as shown in Table 10–2.

## 3.2.7.4.14 Major Frame Identification

Word 72 of minor frames 96 through 103 of the second, third, and fourth PCD major Frames will contain the unique identifier, "1", "2", "3" as in Table 14. The time code is in major frame zero (0).

TABLE 14. PCD MAJOR FRAME IDENTIFIER

PCD MAJOR FRAME	8-BIT IDENTIFIER
1	(msb) 0000 0001 (lsb)
2	0000 0010
3	0000 0011

## 3.2.7.4.15 Attitude Control System Mode

The attitude control system mode is contained in word 72 of minor frame 84 in the fourth major frame of each PCD cycle. This data provides an indication of the CPU mode which is an indication of whether or not the spacecraft attitude is being precision controlled to within the 0.05 degree pointing accuracy. Table 15 shows the CPU modes and word values. The msb is transmitted first. These modes will be described in the following paragraphs.

TABLE 15. ATTITUDE CONTROL SYSTEM MODES

WORD	CPU MODE
(msb) 00000010 (lsb)	Precision (Primary)
(msb) 00001000 (lsb)	Yaw Gyro Compassing (YGC) (Backup)

#### **3.2.7.4.15.1 Precision** (**Primary**)

The computer supplying the PCD is controlling the spacecraft attitude. All ephemeris, attitude, and gyro drift data represent the desired 0.05 degree pointing accuracy.

#### 3.2.7.4.15.2 YGC (Backup)

The computer supplying the PCD is controlling the spacecraft attitude. Attitude and gyro drift data represent 0.2 degree pointing accuracy in pitch and roll and 1.5 degrees in yaw. The spacecraft attitude is not being precision controlled.

# 3.2.7.4.16 *ETM*+ *On/Off Times*

The PCD contains the Last ETM+ On Time and the Last ETM+ Off Time parameters. One of these times is updated each time an On or Off transition is executed by the ETM+. When the ETM+ power supply 1 (A or B) current is greater than or equal to 0.4 amperes and power supply 2 (A or B) current is greater than or equal to 0.4 amperes, the ETM+ is on, the Last ETM+ On time stamp is updated and the data appears in ETM+ major frame (0) word 72, minor frame 42 through 47. When the ETM+ power supply 1 (A and B) current is less than 0.4 amperes or power supply 2 (A and B) current is less than 0.4 amperes, the ETM+ is off, the Last ETM+ Off Time stamp is updated and the data appears in ETM+ major frame (0) word 72, minor frame 84 through 89. The bit pattern for the ETM+ On and Off time code is identical to that in 3.2.7.4.6. The spacecraft controls processor monitors the ETM+ power supply current telemetry every 4 seconds at the 4,096 bps narrowband telemetry rate. Flight software checks for a change in the On/Off state of the ETM+ every PDF major frame (every 4.096 seconds). ETM+ On/Off times are sent in PCD once per PDF cycle (every 16.384 seconds) in major frame (0).

# 3.2.7.4.17 Gyro Select Data

The PCD subcom word 72 of major frame 0, location word 34 contains gyro channel select data (XA, XB, YA, YB, ZA and XB) in bits 0, 1 and 2. These bits indicate the gyro data counts that are provided in the PCD by the PDF. The remaining five bits are not used. The bit values and meanings are in Table 16.

Bit	Value	Information
0 (msb)	1	XA Gyro Selected
0	0	XB Gyro Selected
1	1	YA Gyro Selected
1	0	YB Gyro Selected
2	1	ZA Gyro Selected
2	0	ZB Gyro Selected
3–7 (Isb)	0's	Zero Fill

TABLE 16. GYRO SELECT BIT CODING

#### 3.2.7.4.18 *ETM*+ *Bands On Flags*

The ETM+ bands on-flags in major frame 2, minor frame 32, word 72, bits 0–6 and major frame 2, minor frame 35, word 72, bit 0 appear in the PCD every 16.384 seconds. Typically these are set prior to the imaging interval. However, if the bands are turned off or on during the imaging interval, the SCP receives the ETM+ band flag data from the S-Band telemetry data formatter at a subcom rate equivalent to 16.384 seconds, and not related to the time code contained in the PCD.

## 3.2.7.5 Subcommutation Word 72

Since word 72 of all PCD Minor Frames contain a variety of data essential to the ground segment image processing, Table 17 through Table 20 and Figure 23 through Figure 26 are provided for illustration. Descriptions of the data contained within this word are contained in the previous paragraphs and in the on-orbit handbook. Minor frames 0 through 95 are supplied from the SCP via the CIU to the PDF, including zero fill, for the four major frames (0, 1, 2 and 3) word 72.

TABLE 17. ETM+ PCD MAJOR FRAME (0) WORD 72

MINOR FRAME	DESCRIPTION
0	Euler Parameter 1 (EPA 1)
1	Euler Parameter 1 (EPA 1)
2	Euler Parameter 1 (EPA 1)
3	Euler Parameter 1 (EPA 1)
4	Euler Parameter 2 (EPA 2)
5	Euler Parameter 2 (EPA 2)
6	Euler Parameter 2 (EPA 2)
7	Euler Parameter 2 (EPA 2)
8	Euler Parameter 3 (EPA 3)
9	Euler Parameter 3 (EPA 3)
10	Euler Parameter 3 (EPA 3)
11	Euler Parameter 3 (EPA 3)
12	Euler Parameter 4 (EPA 4)
13	Euler Parameter 4 (EPA 4)
14	Euler Parameter 4 (EPA 4)
15	Euler Parameter 4 (EPA 4)
16	Theta BX
17	Theta BX
18	Theta BX
19	Theta BX
20	Theta BY
21	Theta BY
22	Theta BY
23	Theta BY
24	Theta BZ
25	Theta BZ
26	Theta BZ
27	Theta BZ
28	First 8 bits of last SV clock update
29	Second 8 bits of last SV clock update

TABLE 17. ETM+ PCD MAJOR FRAME (0) WORD 72 (CONT)

MINOR FRAME	DESCRIPTION
30	Third 8 bits of last SV clock update
31	Fourth 8 bits of last SV clock update
32	Fifth 8 bits of last SV clock update
33	Sixth 8 bits of last SV clock update
34	Gyro Select Data
35	Zero Fill
36–41	SV Clock Drift
42	First 8 bits of the time of the last ETM+On Time
43	Second 8 bits of the time of the last ETM+On Time
44	Third 8 bits of the time of the last ETM+On Time
45	Fourth 8 bits of the time of the last ETM+On Time
46	Fifth 8 bits of the time of the last ETM+On Time
47	Sixth 8 bits of the time of the last ETM+On Time
48–49	Zero Fill
50	X – Position
51	X – Position
52	X – Position
53	X – Position
54	Y – Position
55	Y – Position
56	Y – Position
57	Y – Position
58	Z – Position
59	Z – Position
60	Z – Position
61	Z – Position
62	X – Velocity
63	X – Velocity
64	X – Velocity
65	X – Velocity
66	Y – Velocity
67	Y – Velocity
68	Y – Velocity
69	Y – Velocity
70	Z – Velocity
71	Z – Velocity
72	Z – Velocity

TABLE 17. ETM+ PCD MAJOR FRAME (0) WORD 72 (CONT)

MINOR FRAME	DESCRIPTION
73	Z – Velocity
74	Black Body Temperature (Isolated)
75	CFPA Heater Current
76	Calibration Shutter Flag Temperature
77	Backup Shutter Flag Temperature
78	Black Body Temperature (Control)
79	Baffle Temperature (Heater)
80	CFPA Control Temperature
81	Mux 1 Electronics Temperature
82	Mux 1 Power Supply Temperature
83	Mux 2 Electronics Temperature
84	First 8 bits of the time of the last ETM+Off Time
85	Second 8 bits of the time of the last ETM+Off Time
86	Third 8 bits of the time of the last ETM+Off Time
87	Fourth 8 bits of the time of the last ETM+Off Time
88	Fifth 8 bits of the time of the last ETM+Off Time
89	Sixth 8 bits of the time of the last ETM+Off Time
90–95	Zero Fill
96	Bits 0-3 = Spacecraft ID; Bits 4-7 = Hundreds of Days
97	Bits 0-3= Tens of Days; Bits 4-7 = Units of Days
98	Bits 0–3= Tens of Hours; Bits 4–7= Units of Hours
99	Bits 0–3 = tens of Minutes; Bits 4–7 = Units of Minutes
100	Bits 0–3 = Tens of Seconds; Bits 4–7 = Units of Seconds
101	Bits 0–3 = Hundreds of Milliseconds; Bits 4–7 = tens of Milliseconds
102	Bits 0–3 = Units of Milliseconds; Bits 4–7 = Fractions of Milliseconds
103	Bits 0–8 = Zeroes
104–107	Zero Fill
108	Bits 0-3 = 0000 ; Bits 4-7 = ADS-X Temp 1
109	ADS-X Temp 1
110	Bits 0-3 = 0000 ; Bits 4-7 = ADS-Y Temp 2
111	ADS-Y Temp 2
112	Bits 0-3 = 0000 ; Bits 4-7 = ADS-Z Temp 3
113	ADS-Z Temp 3
114	Bits 0-3 = 0000 ; Bits 4-7 = A/D Temp - ADS Electronics
115	A/D Temp – ADS Electronics
116–117	A/D Ground Reference
118–127	Zero Fill

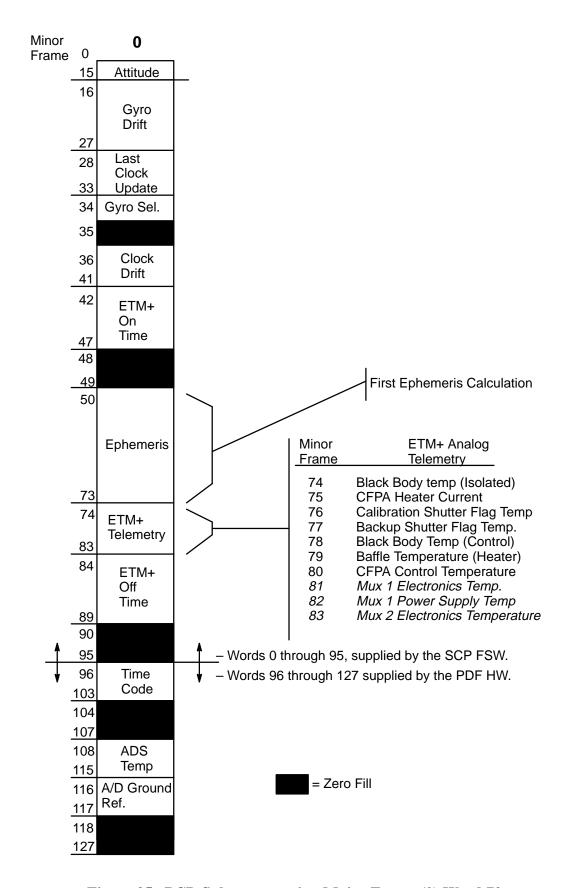


Figure 25. PCD Subcommutation Major Frame (0) Word 72

TABLE 18. ETM+ PCD MAJOR FRAME (1) WORD 72

MINOR FRAME	DESCRIPTION
0	Euler Parameter 1 (EPA 1)
1	Euler Parameter 1 (EPA 1)
2	Euler Parameter 1 (EPA 1)
3	Euler Parameter 1 (EPA 1)
4	Euler Parameter 2 (EPA 2)
5	Euler Parameter 2 (EPA 2)
6	Euler Parameter 2 (EPA 2)
7	Euler Parameter 2 (EPA 2)
8	Euler Parameter 3 (EPA 3)
9	Euler Parameter 3 (EPA 3)
10	Euler Parameter 3 (EPA 3)
11	Euler Parameter 3 (EPA 3)
12	Euler Parameter 4 (EPA 4)
13	Euler Parameter 4 (EPA 4)
14	Euler Parameter 4 (EPA 4)
15	Euler Parameter 4 (EPA 4)
16	X-Position
17	X-Position
18	X-Position
19	X-Position
20	Y-Position
21	Y-Position
22	Y-Position
23	Y-Position
24	Z-Position
25	Z-Position
26	Z-Position
27	Z-Position
28	X-Velocity
29	X-Velocity
30	X-Velocity
31	X-Velocity
32	Y-Velocity
33	Y-Velocity
34	Y-Velocity
35	Y-Velocity
36	Z-Velocity

TABLE 18. ETM+ PCD MAJOR FRAME (1) WORD 72 (CONT)

MINOR FRAME	DESCRIPTION
37	Z-Velocity
38	Z-Velocity
39	Z-Velocity
40–73	Zero Fill
74	Black Body Temperature (Isolated)
75	CFPA Heater Current
76	Calibration Shutter Flag Temperature
77	Backup Shutter Flag Temperature
78	Black Body Temperature (Control)
79	Baffle Temperature (Heater)
80	CFPA Control Temperature
81	Mux 2 Power Supply Temperature
82	Serial Word "J"         Bit           AEM Multiplexer 1 BAND 1 Gain State         0           AEM Multiplexer 1 BAND 2 Gain State         1           AEM Multiplexer 1 BAND 3 Gain State         2           AEM Multiplexer 1 BAND 4 Gain State         3           AEM Multiplexer 1 BAND 5 Gain State         4           AEM Multiplexer 1 BAND 6 PRI Gain State         5           AEM Multiplexer 1 BAND 7 Gain State         6           AEM Multiplexer 1 BAND P Gain State         7
83	Serial Word "K"         Bit           AEM Multiplexer 2 BAND 1 Gain State         0           AEM Multiplexer 2 BAND 2 Gain State         1           AEM Multiplexer 2 BAND 3 Gain State         2           AEM Multiplexer 2 BAND 4 Gain State         3           AEM Multiplexer 2 BAND 5 Gain State         4           AEM Multiplexer 2 BAND 6 PRI Gain State         5           AEM Multiplexer 2 BAND 7 Gain State         6           AEM Multiplexer 2 BAND P Gain State         7
84–95	Zero Fill
96	0000001
97	0000001
98	0000001
99	0000001
100	0000001
101	0000001
102	0000001
103	0000001
104–107	Zero Fill
108	Bits 0-3 = 0000 ; Bits 4-7 = ADS-X Temp1
109	ADS-X Temp 1
110	Bits 0-3 = 0000 ; Bits 4-7 = ADS -Y Temp2

# TABLE 18. ETM+ PCD MAJOR FRAME (1) WORD 72 (CONT)

MINOR FRAME	DESCRIPTION
112	Bits 0–3 = 0000 ; Bits 4–7 = ADS–z Temp 3
113	ADS-Z Temp 3
114	Bits 0-3 = 0000; Bits 4-7 = A/D Temp - ADS Electronics
115	A/D Temp – ADS Electronics
116–117	A/D Ground Reference
118–127	Zero Fill

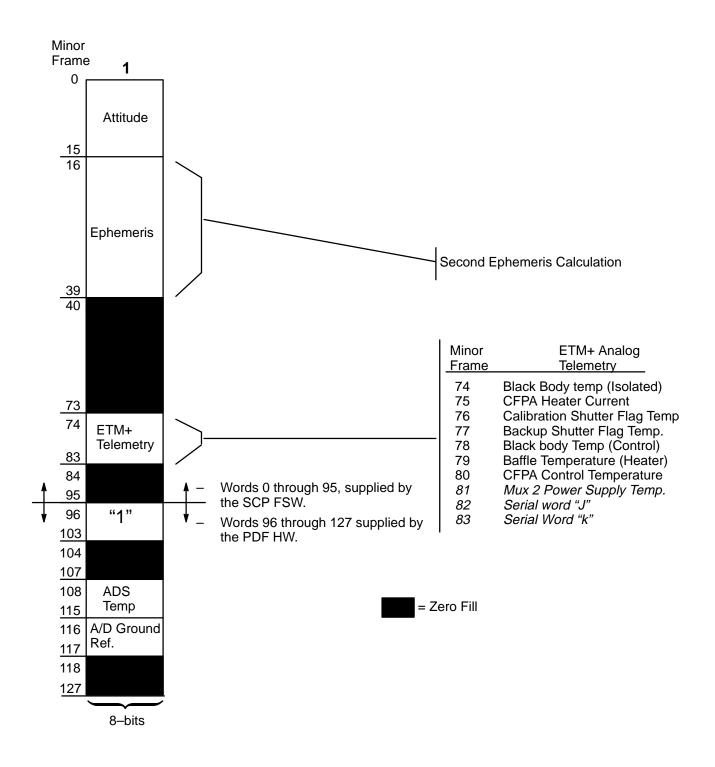


Figure 26. PCD Subcommutation Major Frame (1) Word 72

TABLE 19. ETM+ PCD MAJOR FRAME (2) WORD 72

MINOR FRAME	DESCRIPTION
0	Euler Parameter 1 (EPA 1)
1	Euler Parameter 1 (EPA 1)
2	Euler Parameter 1 (EPA 1)
3	Euler Parameter 1 (EPA 1)
4	Euler Parameter 2 (EPA 2)
5	Euler Parameter 2 (EPA 2)
6	Euler Parameter 2 (EPA 2)
7	Euler Parameter 2 (EPA 2)
8	Euler Parameter 3 (EPA 3)
9	Euler Parameter 3 (EPA 3)
10	Euler Parameter 3 (EPA 3)
11	Euler Parameter 3 (EPA 3)
12	Euler Parameter 4 (EPA 4)
13	Euler Parameter 4 (EPA 4)
14	Euler Parameter 4 (EPA 4)
15	Euler Parameter 4 (EPA 4)
16	MEM Heat Sink Power Supply #1
17	Silicon Focal-Plane Assembly Temp.
18	Zero Fill
19	Baffle Temperature (Tube)
20	MEM Heat Sink Power Supply #2
21	Cold FPA Monitor Temperature
22	Baffle Temperature (Support)
23	Calibration Lamp Housing Temperature
24	Scan-Line Corrector Temperature
25	Calibration Shutter Hub Temperature
26	Ambient Preamp Temperature (High Channels)
27	Band 4 Post Amp Temperature
28	Spare Zero Fill
29	Band 7 Preamp Temperature
30	Ambient Preamp Temperature (Low Channels)
31	Serial Word "A"         Bits           PS 2 Thermal Shutdown Enabled         0           PS 1 Thermal Shutdown Enabled         1           SMA +Z Heater Controller ON         2           SMA -Z Heater Controller ON         3           Spare         4           Shutter Fusible LInk Switch A Closed         5           Shutter Fusible LInk Switch B Closed         6           Shutter Fusible LInk Switch C Closed         7

TABLE 19. ETM+ PCD MAJOR FRAME (2) WORD 72 (CONT)

MINOR FRAME	DESCRIPTION	
32	Serial Word "B"         Bits           Band 1 ON         0           Band 2 ON         1           Band 3 ON         2           Band 4 ON         3           Band 5 ON         4           Band 6/MIR ON         5           Band 7 ON         6           Cold Stage Telemetry ON         7	
33	Serial Word "C"         Bits           Cooler Door Closed         0           Cooler Door Outgas Position         1           Cooler Door Full Open         2           Cooler Door Magnet ON         3           Cooler Door Motor Drive ON         4           Cooler Door Fusible Link Switch A Closed         5           Cooler Door Fusible Link Switch B Closed         6           Cooler Door Fusible Link Switch C Closed         7	
34	Serial Word "D"         Bits           Calibration Lamp 1 ON         0           Calibration Lamp 2 ON         1           Spare         2           Calibration Lamp 1 Backup ON         3           Calibration Lamp 2 Backup ON         4           Spare         5           Spare         6           Spare         7	
35	Serial Word "E"         Bits           Band P ON         0           Spare         1           Blackbody Heater Controller ON         2           Blackbody T2 ON         3           Blackbody T3 ON         4           Blackbody Backup ON         5           SME 1 ON         6           SME 2 ON         7	
36	Serial Word "F"         Bits           Baffle Heater Controller ON         0           Baffle Heater Backup ON         1           Spare         2           Spare         3           Spare         4           Spare         5           Spare         6           Spare         7	
37	Serial Word "G"         Bits           Scan Line Corrector 1 ON         0           Scan LIne Corrector 2 ON         1           Calibration Shutter ON         2           Calibration Shutter Phase Error         3           Calibration Shutter Amplitude Error         4           Backup Shutter ON         5           Backup Shutter Phase Error         6           Backup Shutter Amplitude Erro         7	

TABLE 19. ETM+ PCD MAJOR FRAME (2) WORD 72 (CONT)

MINOR FRAME	DESCRIPTION
38	Serial Word "H"         Bits           Cold Stage Heater Controller ON         0           Cold Stage Outgas Heater Enabled         1           Intermediate Stage Heater Controller ON         2           Intermediate Stage Heater Enabled         3           CFPA Heater Controller ON         4           CFPA T2 Relay ON         5           CFPA T3 Relay ON         6           CFPA Telemetry ON         7
39	Serial Word "I"         Bits           DC Restore Normal         0           Frame DC Restore Selected         1           Telemetry Scaling ON         2           SMA +Z Heater Enabled         3           SMA -Z Heater Enabled         4           Spare         5           SME 1 Select SAM         6           Spare Opto         7
40	Primary Mirror Temperature
41	Primary Mirror Mask Temperature
42	Secondary Mirror Temperature
43	Secondary Mirror Mask Temperature
44	Telescope Housing Temperature
45	Telescope Baseplate Temperature
46	Pan Band Post Amplifier Temperature
47–49	Zero Fill
50	X-Position
51	X-Position
52	X-Position
53	X-Position
54	Y-Position
55	Y-Position
56	Y-Position
57	Y-Position
58	Z-Position
59	Z-Position
60	Z-Position
61	Z-Position
62	X-Velocity
63	X-Velocity
64	X-Velocity
65	X-Velocity
66	Y-Velocity
67	Y-Velocity

TABLE 19. ETM+ PCDMAJOR FRAME (2) WORD 72 (CONT)

MINOR FRAME	DESCRIPTION
68	Y-Velocity
69	Y-Velocity
70	Z-Velocity
71	Z-Velocity
72	Z-Velocity
73	Z-Velocity
74	Black body Temperature (Isolated)
75	CFPA Heater Current
76	Calibration Shutter Flag Temperature
77	Backup Shutter Flag Temperature
78	Black Body Temperature (Control)
79	Baffle Temperature (Heater)
80	CFPA Control Temperature
81	Serial Word "L"         Bit           Cooler Door Direction (1=Open)         0           Cooler Door Move Enable         1           FAC Failsafe Static Motor Power ON         2           FAC Primary Static Motor Power ON         3           FAC Primary Motor Power ON         4           FAC Failsafe Motor Power ON         5           FAC Primary Controller Direction         6           FAC Failsafe Controler Direction         7
82	Serial Word "M"BitMux 1/2 Analog Power Selected0Mux 1/2 Digital Power Selected1Spare2Spare3FAC Primary Controller Single-Step Sizes4FAC Failsafe Controller single-Step Sizes5FAC Primary Controller Power ON6FAC Failsafe Controller Power ON7
83	Serial Word "N"         Bits           AEM Multiplexer 1 ON         0           AEM Multiplexer 2 ON         1           AEM Multiplexer 1 MDE ON Status         2           AEM Multiplexer 2 MDE ON Status         3           AEM Multiplexer 1 Band 6 RDT Gain State         4           AEM Multiplexer 2 Band 6 RDT Gain State         5           AEM Multiplexer 1 Data Priority Selected         6           AEM Multiplexer 2 Data Priority Selected         7
84	Serial Word "P"         Bit           FAC Stow Position Switch PRI         0           FAC Stow Position Switch RDT         1           FAC Calibration Position Switch PRI         2           FAC Calibration Position Switch RDT         3           FAC Calibration/Stow Move ON Status PRI         4           FAC Calibration/Stow Move ON Status RDT         5           FAC Single-Step Move ON Status PRI         6           FAC Single-Step Move ON Status RDT         7
85–95	Zero Fill
96	0000010

TABLE 19. ETM+ PCDMAJOR FRAME (2) WORD 72 (CONT)

MINOR FRAME	DESCRIPTION
97	0000010
98	0000010
99	0000010
100	0000010
101	0000010
102	0000010
103	0000010
104–107	Zero Fill
108	Bits 0-3 = 0000 ; Bits 4-7 = ADS-X Temp 1
109	ADS Temp 1
110	Bits 0-3 = 0000 ; Bits 4-7 = ADS-Y Temp 2
111	Ads-Y Temp 2
112	Bits 0–3 = 0000; Bits 4–7 = ADS–Z Temp 3
113	ADS-Z Temp 3
114	Bits 0-3 = 0000 ; Bits 4-7 = A/D Temp - ADS Electronics
115	A/D Temp – ADS Electronics
116–117	A/D Ground Reference
118–127	Zero Fill

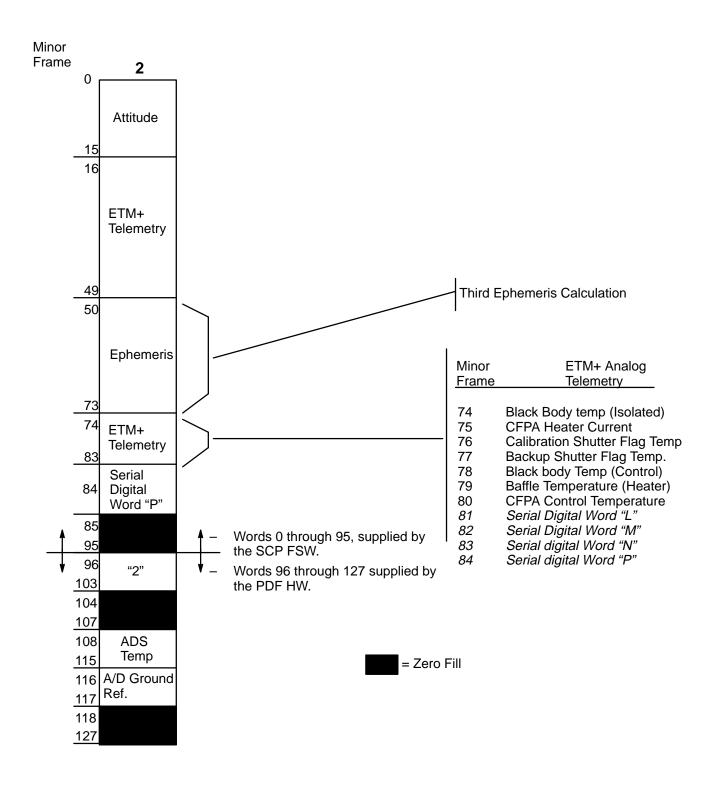


Figure 27. PCD Subcommutation Major Frame (2) Word 72

TABLE 20. ETM+ PCD MAJOR FRAME (3) WORD 72

MINOR FRAME	DESCRIPTION
0	Euler Parameter 1 (EPA 1)
1	Euler Parameter 1 (EPA 1)
2	Euler Parameter 1 (EPA 1)
3	Euler Parameter 1 (EPA 1)
4	Euler Parameter 2 (EPA 2)
5	Euler Parameter 2 (EPA 2)
6	Euler Parameter 2 (EPA 2)
7	Euler Parameter 2 (EPA 2)
8	Euler Parameter 3 (EPA 3)
9	Euler Parameter 3 (EPA 3)
10	Euler Parameter 3 (EPA 3)
11	Euler Parameter 3 (EPA 3)
12	Euler Parameter 4 (EPA 4)
13	Euler Parameter 4 (EPA 4)
14	Euler Parameter 4 (EPA 4)
15	Euler Parameter 4 (EPA 4)
16	X-Position
17	X-Position
18	X-Position
19	X-Position
20	Y-Position
21	Y-Position
22	Y-Position
23	Y-Position
24	Z-Position
25	Z-Position
26	Z-Position
27	Z-Position
28	X-Velocity
29	X-Velocity
30	X-Velocity
31	X-Velocity
32	Y-Velocity
33	Y-Velocity
34	Y-Velocity
35	Y-Velocity
36	Z-Velocity

TABLE 20. ETM+ PCD MAJOR FRAME (3) WORD 72 (CONT)

MINOR FRAME	DESCRIPTION
37	Z-Velocity
38	Z-Velocity
39	Z-Velocity
40–73	Zero Fill
74	Black Body Temperature (Isolated)
75	CFPA Heater Current
76	Calibration Shutter Flag
77	Backup Shutter Flag Temperature
78	Black Body temperature (Control)
79	Baffle Temperature
80	CFPA Control Temperature
81	Serial Word "Q" Bits FAC Pull-Pin Heater 1 ON 0 FAC Pull-Pin Heater 2 ON 1 FAC Pull-Pin Heater 2 ON 1 FAC Pull-Pin Heater Power, Enable PRI 2 FAC Pull-Pin Heater Power 3 FAC Pull-Pin Retracted Position Switch PRI 4 FAC Pull-Pin Retracted Position Switch RDT 5 FAC Pull-Pin Fully Retracted Position Switch PRI 6 FAC Pull-Pin Fully Retracted Position Switch RDT 7
82	Serial Word "R":BitsFAC Primary CW Rotation Switch Status0FAC Primary CCW Rotation Switch Status1FAC Redundant CW Rotation Switch Status2FAC Redundant CCW Rotation Switch Status3Spare4Spare5Spare6Spare7
83	Serial Word "S" Bits
	Command Reject, Enable 1 P 0 Command Reject, Enable 2 P 1 Command Reject, Enable 3 P 2 Command Reject, Enable 4 P 3 Command Reject, Enable 1 R 4 Command Reject, Enable 2 R 5 Command Reject, Enable 3 R 6 Command Reject, Enable 4 R 7
84	Spacecraft CPUMODE
85–95	Zero Fill
96	00000011
97	00000011
98	00000011

TABLE 20. ETM+ PCD MAJOR FRAME (3) WORD 72 (CONT)

MINOR FRAME	DESCRIPTION
99	00000011
100	00000011
101	00000011
102	00000011
103	00000011
104–107	Zero Fill
108	Bits 0-3 = 0000 ; Bits 4-7 = ADS-X Temp 1
109	ADS-X Temp 1
110	Bits 0-3 = 0000 ; Bits 4-7 = ADS-Y Temp 2
111	ADS-Y Temp 2
112	Bits 0-3 = 0000 ; Bits 4-7 = ADS-Z Temp 3
113	ADS-Z Temp 3
114	Bits 0-3 = 0000 ; Bits 4-7 = A/D Temp - ADS Electronics
115	A/D Temp – ADS Electronics
116–117	A/D Ground Reference
118–127	Zero Fill

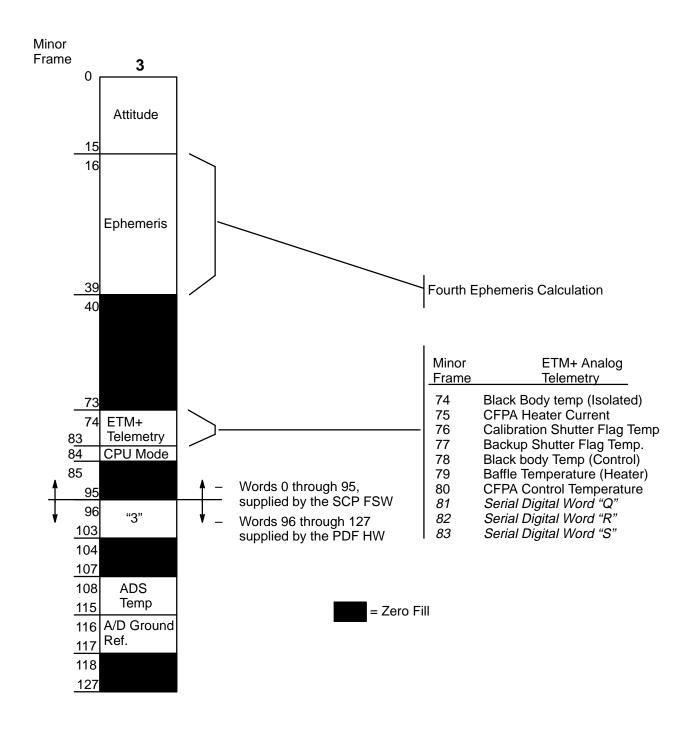


Figure 28. PCD Subcommutation Major Frame (3) Word 72

#### 3.3 SOLID STATE RECORDER

#### 3.3.1 FUNCTIONAL ARCHITECTURE

Figure 29 shows the functional architecture and relationship between the ETM+ instrument, the Solid State Recorder (SSR), and the X-band communication links. ETM+ wideband data is collected by the ETM+ sensors and CCSDS-encoded by formatters internal to the ETM+. The CCSDS-formatted data is transmitted by the ETM+, in two 75 Mbps bitstreams, to the Baseband Switching Unit (BSU) for immediate transmission via an X-band link and/or for recording on the SSR. The two 75 Mbps ETM+ bitstreams remain a matched pair throughout record, playback and transmission operations. The wideband data is recorded at an aggregate rate of 150 Mbps. For transmission of recorded wideband data, the recorded data is played back from the recorder using one or two 150 Mbps bitstreams and sent to the X-band modulator via the BSU.

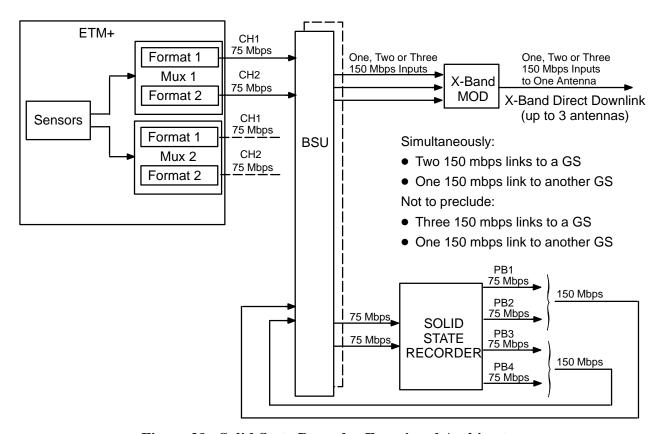


Figure 29. Solid State Recorder Functional Architecture

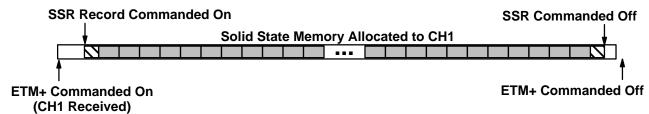
#### 3.3.2 RECORD FORMAT

The solid state recorders will record ETM+ CADU data as two bitstreams, each at a nominal rate of 74.914 Mbps. CADU data is recorded in the same order as received from the ETM+. Partial CADUs may be recorded if the ETM+ collection interval extends prior to or beyond the commanded SSR record interval. An example of a record format is shown in Figure 30.

#### 3.3.3 PLAYBACK FORMAT

The SSR playback data is read out of memory and mapped into one or two groups of two 75 Mbps bitstreams with a total aggregate rate of 150 or 300 Mbps. The bitstreams consist of the data generated

# ETM+ FMT1 (CH1) Recording:



#### ETM+ FMT2 (CH2) Recording:

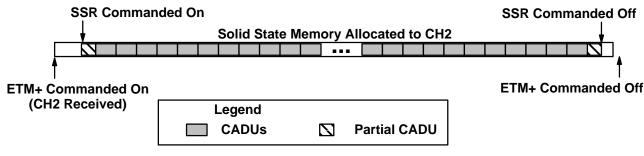


Figure 30. Record Format Example

by the ETM+. Record intervals, each corresponding to a ETM+ collection interval consisting of one or more Landsat scenes, may be subdivided for playback if more than one scene is collected. In this case, each resulting subinterval is defined such that data in the vicinity of each subinterval boundary are included (redundantly) within both subintervals. Each subinterval includes all of the CADU data required to process the subinterval as a separate ETM+ collection. As a result, individual subintervals may contain partial CADUs such as that described in paragraph 3.3.2. An example of the SSR playback is shown in Figure 31. A PN code may precede the SSR playback data for ground station syncing.

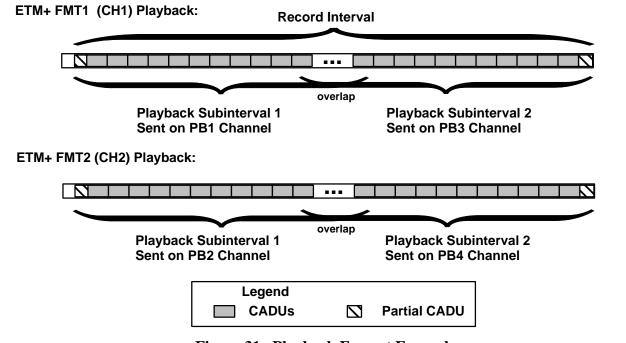


Figure 31. Playback Format Example

# APPENDIX A ACRONYM LIST

A/D Analog/Digital

ADA Attitude Displacement Assembly

ADS Attitude Displacement Sensors

AEM Auxiliary Electronics Module

BCD Binary-Coded Decimal

BCH Bose–Chaudhuri–Hocquenghem

BD Band

BSU Baseband Switching Unit

CADU Channel Access Data Unit

CCSDS Consultative Committee for Space Data Systems

CFPA Cold Focal Plane Array

CRC Cyclic Redundancy Check

DIR Direction

ECI Earth Centered Inertial

ECITOD Earth Centered Inertial True of Date

EDAC Error Detection and Correction

EOL End-of-Line

EPA Euler Parameter

ERR Error

ETM+ Enhanced Thematic Mapper Plus

FHS First Half Scan

FMT Format

HDR Header

Hex Hexadecimal

ICD Interface Control Document

IFOV Instantaneous Field of View

IGS International Ground Stations

lsb Least Significant Bit

IMU Inertial Measurement Unit

LSB Least Significant Bit

LWIR Thermal Long Wavelength Infrared

MBPS Mega Bits Per Second

msb Most Significant Bit

MSB Most Significant Byte

MF Minor Frame

OSI Open System Interconnection

PAN Panchromatic

PCA Physical Channel Access

PCD Payload Correction Data

PCSS Program Coordinate Systems Standards

PFPA Prime Focal Plan Assembly

PN Pseudo-Random Noise

PNTR Pointer

TBD To Be Determined

TBR To Be Resolved

SCID Spacecraft Identifier

SCN Scan

SD Scan Direction

SHS Second Half Scan

SIG Signalling

SLL Scan Line Length

SD Scan Direction

SLS Scan Line Start

SV Space Vehicle

SWIR Short Wavelength Infrared

VCA Virtual Channel Access

VCDU Virtual Channel Data Unit

VCID Virtual Channel Identifier

VCLC Virtual Channel Link Control

VER Version

VNIR Visible and Near Infrared

YGC Yaw Gyro Compassing

# APPENDIX B REED-SOLOMON CODING

## **B.1 SPECIFICATION**

The parameters of the selected Reed-Solomon (R\_S) code are as follows:

- (1) J=4 bits per R–S symbol.
- (2) E= 2 R-S symbols error correction capability within a Reed-Solomon code word.
- (3) General characteristics of Reed–Solomon codes:
  - (a) J,E, and I (the depth of interleaving) are independent parameters.
  - (b)  $n=2^{J}-1=15$  symbols among n symbols of an R-S codeword.
  - (c) 2E is the number of R–S symbols among n symbols of an R–S codeword representing check symbols.
  - (d) k = n-2E is the number of R-S symbols among n R-S symbols of an R-S codeword representing information.
- (4) Field generator polynominal:

$$F(x) = x^4 + x + 1$$
  
over  $GF(2)$ .

(5) Code generator polynominal:

$$g(x) = \prod_{j=6}^{9} (x-a^{j}) = \sum_{i=0}^{4} G_{i}x^{i}$$

over  $GF(2^4)$ .

(6) It should be recognized that F(x) and g(x) characterize a (15,11) Reed–Solomon code. But since there are not 11 bytes in the header to encode, the R–S code was shortened to a (10,6) RS code. This implies that there are 5 virtual fill symbols. Also note that there is no interleaving.

# B.2 GALOIS FIELD TABLE FOR $GF(2^{\Lambda}4)$ GENERATED BY F(X) FROM NUMBER 4 ABOVE

P O W E R	POLY IN ALPHA
*	0000
1	0001
2	0100
3	1000
4	0011
5	0110
6	1100
7	1011
8	0101
9	1010
10	0111
11	1110
12	1111
13	1101
14	1001

# **B.3** EXPANSION OF REED-SOLOMON COFFECIENTS

COEFFICIENTS OF g(x)	POLYNOMINAL IN a
	a <sup>3</sup> a <sup>2</sup> a <sup>1</sup> a <sup>0</sup>
$G_0 = G_4 = a^0$	0 0 0 1
$G_1 = G_3 = a^3$	1 0 0 0